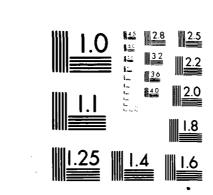
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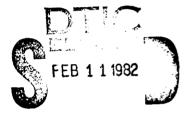
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DOCUMENTATION FOR SWATH SHIP RESISTANCE
OPTIMIZATION PROGRAM (SWATHO) USER'S AND
MAINTENANCE MANUAL

Toby J. Nagle

and

Arthur M. Reed



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SHIP PERFORMANCE DEPARTMENT

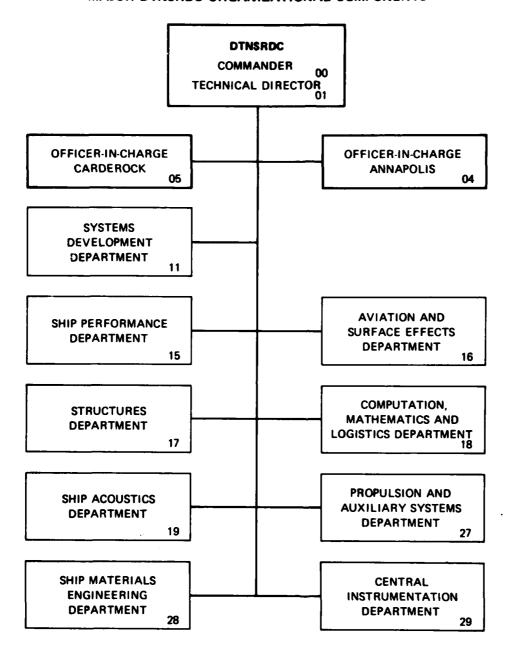
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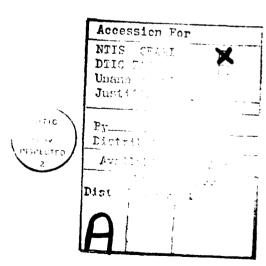
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NOMENCLATURE

Symbol	Description
A(x)	Body sectional area curve
A	Maximum section area of body
A _{Bm}	Symmetric coefficients of Chebychev series for body
A _{Sm}	Symmetric coefficients of Chebychev series for strut
A_{W}	Area of the waterplane
Ъ	Half of the separation distance of the hulls
B _{Bm}	Anti-symmetric coefficients of Chebychev series for body
B _{Sm}	Anti-symmetric coefficients of Chebychev series for strut
C _A	Correlation allowance
c _F	Frictional resistance coefficient
С _{Fm}	Form drag coefficient
$^{\mathrm{C}}_{\mathrm{P}}$	Body prismatic coefficient
$^{\mathrm{C}}$ WP	Waterplane coefficient
c_{IW}	Waterplane longitudinal inertia coefficient
g	Acceleration due to gravity
h _B	Maximum depth of submergence of axis of body
h _S	Maximum draft of strut
$J_{n}(\alpha)$	Bessel functions

NOMENCLATURE (Continued)

Symbol	Description
$^{L}_{B}$	Maximum length of body
^L s	Maximum length of strut
$^{ m P}_{ m E}$	Effective power
R _n	Reynolds number
$^{R}_{T}$	Total ship resistance
$^{R}W_{B}$	Wave resistance due to one main body
$^{\mathrm{R}}_{\mathrm{F}}$	Frictional resistance
$^{R}\mathrm{F_{m}}$	Form drag
$^{R}W_{S}$	Wave resistance due to one strut
$^{R}_{W_{SB}}$	Wave resistance due to the interaction of strut and main body
$R_{\widetilde{W}}$	Wave resistance
S	Wetted surface
T (T also used)	Thickness at mid length of strut
t(x)	Strut half thickness function
T _B mn	Auxiliary wave resistance function used in calculation of ${\bf R}_{\widetilde{\bf W}_{B}}$
T _S mn	Auxiliary wave resistance function used in calculation of ${\rm R}_{\widetilde{W}_{S}}$

NOMENCLATURE (Continued)

Symbol Symbol	Description
TSB mn	Auxiliary wave resistance function used in calculation of ${\tt R}_{{\tt WSB}}$
U _m (x)	Chebychev cosine series term (symmetric)
V	Velocity of ship
V _m (x)	Chebychev sine series term (asymmetric)
W _B mn	Auxiliary wave resistance function used in calculation of $\boldsymbol{R}_{\boldsymbol{W}_{\boldsymbol{B}}}$
W _{S_{mn}}	Auxiliary wave resistance function used in calculation of $R_{\begin{subarray}{c}W\\S\end{subarray}}$
W _{SB} _{mn}	Auxiliary wave resistance function used in calculation of $^{R}_{\mathrm{SB}}$
α	Variable of integration
ρ	Density of water
Yob	Dimensionless wave number related to body length and ship speed
Yos	Dimensionless wave number related to strut length and ship speed
ζ	Variable of integration

ENGLISH/SI EQUIVALENTS

1 degree (angle) = 0.01745 rad (radians)
1 foot = 0.3048 m (meters)

1 foot = 0.3048 m (meters) 1 foot per second (fps) = 0.3048 m/sec (meters per second)

1 inch = 25.40 mm (millimeters)

1 knot = 0.5144 m/s (meters per second)

1 1b (force) = 4.448 N (Newtons)

1 inch-lb - (Inch 1bs) = 0.1130 N·m (Newton-meter)

1 long ton (2240 pounds) = 1.016 metric tons; or 1016 kilograms

1 horsepower = 0.746 kW (kilowatts)

ABSTRACT

This report documents a computer program, SWATHO, which optimizes SWATH ship geometry, resulting in a minimum resistance and EHP for a specified speed of the ship. Contained in the report is a detailed description of the procedure for assembling the input deck for use on the CDC-6000 computer system at DTNSRDC, running the program on this system and understanding the output from the program. Also included are main program and subroutine descriptions as well as the theory behind the computations being performed.

ADMINISTRATIVE INFORMATION

This investigation was authorized under a direct funded block from the Naval Material Command (NAVMAT 08T2) under Program Element 62543N, Task Area ZF43-421-001, and administered by the David W. Taylor Naval Ship Research and Development Center (DTNSRDC), Work Unit 1500-102-30, and 1-1500-104-70.

INTRODUCTION

The computer program SWATHO, developed at DTNSRDC, optimizes SWATH ship geometry to arrive at minimum values of resistance and effective horsepower for a predetermined speed. The minimum values arrived at by the optimization are a result of initial ship geometry and constraints on the geometry as specified by the program user.

The calculations use standard EHP prediction methods accounting for frictional resistance, form drag, and wave resistance. Wave making predictions are made using the work of Lin and Day 1 on twin hull ships.

Contained in this report is a user manual which explains how to assemble the input deck as well as how to run the program on the CDC-6000 computer

¹References are listed on page 100

system at DTNSRDC. A description of the output and how to interpret it is also included along with an overview of how the main program works. Following the user's manual, a maintenance manual is presented, which contains a detailed description of the common blocks, functions and subroutines, and a tree diagram. The appendices contain a description of the objective and penalty functions used for minimization of effective power, a presentation of the theory behind the computational procedures for the resistance predictions, and an explanation of the relationships between Chebychev series and SWATH hullform coefficients.

This program will assist naval architects in finding a minimum resistance hull design which satisfies the given geometric constraints. The User's Manual portion of this document is intended to provide the user with the information needed to prepare input for the program, and with a description of the resulting output. The Maintenance Manual provides the documentation needed to understand the functioning of the program and its various subprograms, at a detailed enough level that changes could be made in the program if necessary.

USER'S MANUAL

PROGRAM DESCRIPTION

The Program SWATHO optimizes a SWATH ship geometry to obtain a minimum value of resistance and EHP for a specified speed of the ship. The program can be better understood if it is divided into four parts, each part having a specific purpose and result.

The first part of the program reads the input data that the user has prepared on cards; thus initializing the ship description as well as setting up the constraints. The array of scaling factors, which is used as the initial geometry, is optimized and initially set to 1 in this part. After each run, the values in the scaling array which correspond to the value of minimum EHP at that point are input to the next run of the program, keeping the rest of the INPUT cards the same. The actual description and format of the INPUT cards are found in the INPUT DESCRIPTION and INPUT EXAMPLE of this USER'S MANUAL.

The second part of the program actually performs the optimization of the EHP function. A detailed description of the method used and the actual subroutines which perform the optimization can be found in the MAINTENANCE MANUAL.

The third part of the program evaluates the objective function, thus finding intermediate values and a final value of EHP. A penalty function is also calculated in this part, based on any violation of constraints by the ship geometry at that time. Then more optimization is performed. This optimization cycle is repeated until the computational time is exceeded or sufficient convergence is reached.

The last part of the program outputs the initial ship description, intermediate curves and a design description, as well as resistance calculations over a range of speeds and other results of calculations. A detailed description of the output is found in the OUTPUT DESCRIPTION and OUTPUT EXAMPLE in the USER'S MANUAL.

INPUT DESCRIPTION

Diagrams showing the orientation of the geometry represented by the variables involved in this input description are found in Figure 1.

INPUT CARDS (Total number of input cards is 8)

1. The first card contains the alphanumeric title of the model.

Variable:

TITLE (8)

Format:

8A10

1	11	21	31	41	51	61	71
TITLE	(1) TITLE	(2) TITLE	(3) TITLE	(4) TITLE	(5) TITLE	(6) TITLE	(7) TITLE (8)

2. Data on the second card are:

XLSI - Length of strut in ft

HSI - Draft of strut in ft

TSMAXI - Strut thickness in ft at XLSI/2

CWPI - Waterplane area coefficient

CLCFI - Waterplane moment coefficient

KG - Height of CG above baseline

Variable: XLSI, HSI, TSMAXI, CWPI, CLCFI, KG

Format: 6F10.6

1 11 21 31 41 51

XLSI | HSI | TSMAXI | CWPI | CLCFI | KG

3. Data on the third card are:

XLBI - Length of body in ft

BDIAI - Body diameter in ft at XLBI/2

BSI - Separation of hull centerlines

CSTRTI - Dist. of strut CL fwd of body CL

CPI - Body prismatic coefficient

CLCBI - Body moment coefficient

Variable: XLBI, BDIAI, BSI, CSTRTI, CPI, CLCBI

Format : 6F10.6

1	11	21	31	41	51	
XLBI	BDIAI	BSI	CSTRTI	CPI	CLCBI	

4-5 Data on the fourth and the fifth card are:

S - Array of scale factors

Variable: S

Format : 8 F10.6

1	.11	21	31	41	51	61	71	
S(1)	S(2)	S(3)	S(4)	S(5)	S(6)	S(7)	S(8)	j

6. Data on the sixth card is:

OPTSPD - Optimization speed in knots

Variable: OPTSPD
Format: F10.6

OPTSPD

7. Data on the seventh card are:

DISPMN - Minimum ship displacement in long tons

DFAC - Minimum draft/body diameter ratio

DRFTMX - Maximum draft of ship

WMAX - Maximum width of ship

XFWD - Minimum distance from body leading edge to strut leading edge

XAFT - Minimum distance from body trailing edge to strut trailing edge

Variable: DISPMN, DFAC, DRFTMAX, WMAX, XFWD, XAFT

Format: 6F10.4

8. Data on the eighth card are:

MINGMT - Minimum TRANSVERSE GM

MINGML - Minimum LONGITUDINAL GM

LODMAX - Maximum body length over diameter ratio

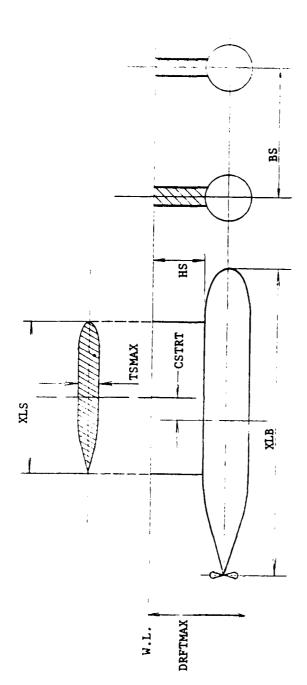
LODMIN - Minimum body length over diameter ratio

TSMIN - Minimum strut thickness at 50 percent chord

Variable: MINGMT, MINGML, LODMAX, LODMIN, TSMIN

Format: 5F10.1

1	11	21	3i	41
MINGMT	MINGML.	LODMAX	LODMIN	TSMIN



SINGLE STRUT SWATH

Figure 1. SWATH SHIP BODY AND STRUT PROFILES (SWATHO)

INPUT EXAMPLE

Example of Input Data (Optimization):

1	11	21	31	41	51	61	71
TITLE (1)	TITLE (2)	TITLE (3)	TITLE (4)	TITLE (5)	TITLE (6)	TITLE (7)	TITLE (8)

SWATH OPTIMIZATION PROGRAM

1	11	21	31	41	51	
XLSI	HSI	TSMAXI	CWPI	CLCFI	KG	7
210.	10.725	6.500	.8511	.014286	20.	

1	11	21	31	41	51
XLBI	BDIAI	BSI	CSTRTI	CPI	CLCBI
260.	14.30	68.50	15.	.849	.01538

1	11	21	31	41	51	61	71
S(1)	S(2)	S(3)	S(4)	\$(5)	S(6)	S(7)	S(8)
1.0154	.9900	0.9900	0.9783	-1.0	.9032	-1.0000	1.005

1	11	21	31
S(9)	S(10)	S(11)	S(12)
.9900	1.2500	1.0000	.5000

OPTSPD 20.

1	11	21	31	41	51
DISPMN	DFAC	DRFTMX	WMAX	XFWD	XAFT
2675.0	1.67	30.0	106.0	7.50	30.0

1	11	21	31	41
MINGMT	MINGML	LODMAX	LODMIN	TSMIN
30.00	30.00	20.00	16.00	5.0

OUTPUT DESCRIPTION

The first page of output consists of an echo print of the input including the S array of scaling factors. Following the array of scaling factors, NL and NF (value of counters in the optimization procedure)

P (the penalty value), PEHP (the function to be minimized), and a repeat of the scaling factors (S array) are printed.

The second page of output consists of an intermediate plot of a body sectional area curve and strut waterplane outline curve where the symbols B and S stand for body and strut respectively. At the bottom of the plot, the values for the strut and body geometry are printed with the effective horsepower required for the design.

The third page of output lists values for intermediate design consisting of the symmetric and antisymmetric Chebychev coefficients for the body and strut and the wetted surface and ship geometry for the optimization speed. Also a table of speed-length ratios, wave resistance, frictional and residual resistance coefficients as well as the actual resistances and the EHP predicted is printed for the range of speeds from 10 to 26 knots.

The fourth page of output consists of a tabular listing of all of the scale factors of the S array corresponding to the strut and body geometric variables, as well as the penalty value and PEHP. The second difference array D(*) is printed out at the bottom of the listing. Also printed is NL, NF, PEHP and the values of the S array corresponding to a minimum PEHP so far. These S values are input by the user for the next run of the program. Often, when the execution time limit is exceeded, the output will stop in the middle of this tabular listing, in which case the minimum PEHP is found and the S array of scale factors are picked out from the table and used as input for the subsequent run.

The sequence of body sectional area curve and strut waterplane curve, intermediate design description, followed by tabular output is repeated until the time limit is exceeded or until convergence of the PEHP function is reached.

When the program has reached a prescribed degree of convergence, a final page is printed out, entitled "Wave Resistance for Strut and Body Geometric Characteristics." This page contains the optimum strut and body geometric characteristics, separation in feet, B/LB, strut centerline distance from body centerline (STRUT OFFSET), strut offset/LE, the wave drag for strut and body and the frictional drag of body and strut. The coefficients of resistance as well as the minimum EHP calculated at certain speeds are also given.

OUTPUT EXAMPLE

SWATH OPTIMIZATION PROSRAM -- SWATH OPTIMIZATION PROSRAM - GML.6NT - AJGUST 1980

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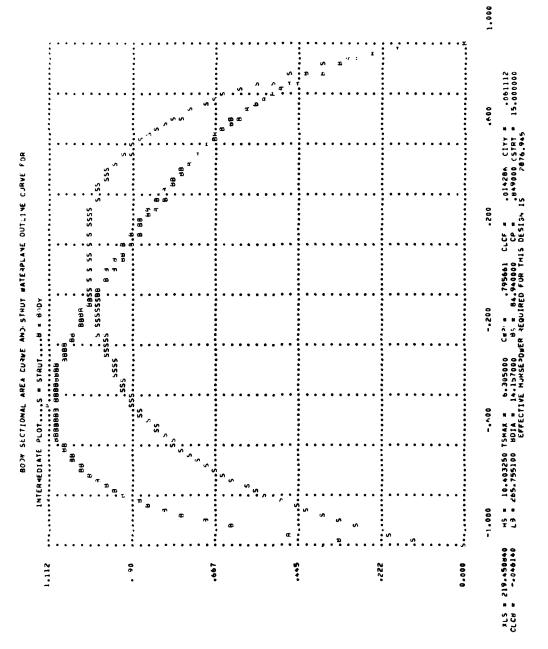
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ERMEDIATE DESIGN STRUT CHEBYCHEV COEFFICIENTS - ASM STRUT CHEBYCHEV COEFFICIENTS - BSM BODY CHEBYCHEV COEFFICIENTS - BSM STRUT WETTED SURFACE IN SQUARE FEET BODY OF STRUT IN FEET BATERPLAME AREA COEFFICIENT = BATERPLAME MOMENT COEFFICIENT = BATERPL	BOODY PRISABLE COEFFICENT BOODY MOMENT CUEFFICIENT SHIP DISPLEMENT IN LONG TOWS SHIP DR FT / BOD DIAMETER DRAFT DF SHIP IN FEFT BREADTH OF SHIP IN FEFT DISTANCE L.E.BOD. TO L.E.STRUT BREAT TR NESTSES GHL (LONGITUDAMA) GHL (LONGITUDAMA) SHIP INCKNESS AT 0.5 CHORD	7
ATE DESIGN GERWCHEV COFFIC CHEBYCHEV COFFIC LANE MOMENTA COFFIC LANE MOMENTA COFFIC CANE MOMENTA COFFIC CAN	CCUEFFI CEMENT CEMENT CEMENT 1P IN F SMIP IN F ERSED UDAMAL) UDAMAL) NESS AT	2000 0000 0000 0000 0000 0000 0000 000
INTERNÉDIATE DESIGN STRUT ÉMERYCHEV COEFFIC STRUT ÉMENCHEV COEFFIC BODY CHÉBYCHEV COEFFIC STRUT METTÉD SURFACE IN BODY MITED SURFACE IN TOTAL METTÉD SURFACE IN TOTAL METTÉD SURFACE IN TOTAL METTÉD SURFACE IN TOTAL METTÉD SURFACE IN MAX. ST. UT THICKNESS IN MATERPLANE AREA COEFFIC MATERPLANE AREA COEFFIC MATERPLANE MOMENT COEFF MATERPLANE MOMENT TEET A SERVICE MOMENT OF COEFF MATERPLANE MOMENT TO DET MOMENT OF COEFF MATERPLANE MOMENT TO DE MOMENT OF COEFF MOMENT TO DE MOMENT TO DE MOMENT OF COEFF MOMENT TO DE MOMENT TO DE MOMENT TO DE MOMENT TO DE MOM	BODY PRISMATIC COEFFICIENT BODY WOMENT CUEFFICIENT SMIP DISPLACEMENT IN LONG SMAFT OF SHIP IN FEET DISTANCE L.E. 800. TO L.E.S. BMT ITR MYSERS. MML LANGITUDANAL) BODY LENGTH / 800Y DIA.ETE STRUT THICKNESS AT 0.5 CHO.	V-F-S 10-878 23-629 27-002 33-36 33-756 33-756 49-503 49-603
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At this point numerous iterations have been omitted for a more concise report presentation.

INTERMEDIATE DESIGN

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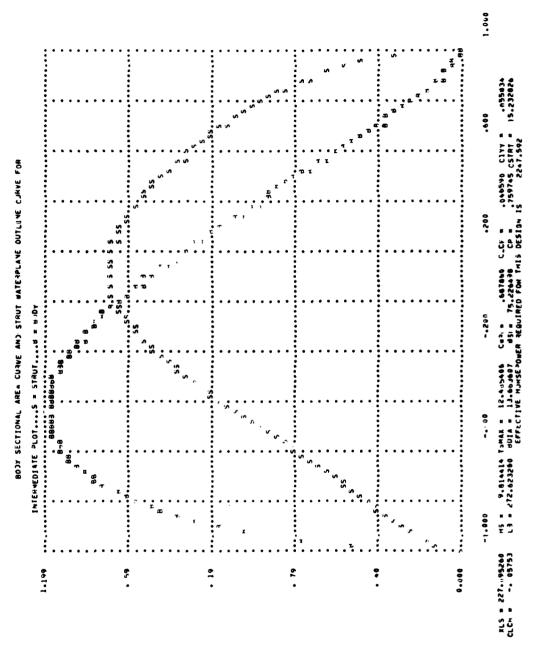
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INTERMEDIATE DESIGN

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SWATH OPTIMIZATION PROSRAM - GAL, GAT - AUGUST 1980

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STRUT CHEBYCHEV COEFFICIENTS +	CHEBYCHEV	CHEBYCHEV	STRUT WETTED SURFACE IN SUJARE	WETTED SURFACE IN	L WETTED SURFACE IN	LENGTH OF STRUT IN FEET	DRAFT OF STRUT IN FEET	MAX. ST UT THICKNESS IN FEET	WATERPLANE AREA COEFFICIENT		WATERPLANE INERTIA COEFFICIENT	LENGTH OF BODY IN FEET	BODY DI METER IN FEET AT X_B/2	SEPARATION OF HUIL CENTERLINES	DIST. STRUT CL F.D OF BUDY CL	BODY PRISMATIC CHEFFICIENT	HODY MOMENT COEFFICIENI	SHIP DI - PLACEMENT IN LONG TONS	SHIP OR FT / BOD. DIAMETER	DRAFT OF SHIP IN FEET	BREADTH OF SHIP IN FEET	DISTANCE L.E. BOD. TO L.E. STRUT	GMT (TR NSVERSE)	GML (LONGITUDANAL)	BODY LENGTH / BONY DIAMETER	STRUT T ICKNESS T 0.5 CHO20

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GML + 647	
•	
PROSRAM	
NC11A1IHITGC	
SWATH	
MAVE RESISTANCE CALCULATIOUS FOR SWATH OPTIMICATION PROGRAM - GML.641 - AUGUST 1990	CHARACTERISTICS
RESISTANCE	GEOMETRIC
MAVE	STRUT

						BONY V-L MATIO =
TSMAX = 12.495406 MS/LS = .043217		Ax = 146.629250 n~/LB = .061059	.270	950° ≖		710 = .66
TSMAX = nS/LS =		AX == 11/18 ==		STHUM OFSETZLA	_	STRUT V-L RATIO =
9.814.14		HB = 16,646217 F30UD5 = .180213	8~/8 97		WETTED SURFACES ARE FOR A DEMI	4N015.
HS #	SOI	M8 ≈ F30U05 =	T = 75.226	.EET = 15.	URFACES ARE	PS. 10.000
227.09.260 12.824521	BODY GEOMETRIC CHARACTEMISTICS	272.621260 15.394578	SEPARATION DISTANCE IN FEET	STRUT CL FROM HODY CL IN FEET = 15.233	#ETTE0 S	SMIP SPEED IS 16.478 FPS. 10.000 KNOTS.
KL S GAMAOS	800' GE(XLB GAMAOB	SEPARA	STRUT (SHIP

RESISTANCES IN POUNDS ARE FUM A DEMI-HULL

.606

		.5756-03			.222E-n3	.797E-04				
		Cwl ToT =			r#2 TuT =	CR TOTAL	•			
.16733 67E -04 .4 8 5E -03	.1~2997 8E .03 .047E-03	.1471757E+ 3 .043E-03	. 3235560E+03 .094E+03	.1159055E+.2 .003c+u3	.+305455E+03 .125E-03		.29051156.0	.642E-03	.490808£ 0.	1.422E-03
, N		, "	. *	. "						
STRUT MAVE DRAG IN POJADS	BODY WAVE URAG IN POUNUS RIB/(RHO/SHWIS)RF4V842)	STRUT-BODY WAVE DRAG IN LBS : 7154/(RM:/2**TSURF*V**2)	STRUT INTERFERENCE DRAG L3S: 3125/(RH://2*/TSURFOVOO2)	BODY INTERFERENCE URAS LAS : ALEM (RM./2**TSUNF*V**2)	STRUT-BOD INTERNCE DAMB _B : 21253/(RHO/2=WISUKFevee2)		STRUT FRICTION DHAG IN LSS	4FS/(AHO/20NTSJRF0V002)	30DY FRICTION DRAG IN LUS	AFR/(RMO/20HTSJRFoveez)

RF TOTAL = .7813200E.04 CF TOTAL = 2.264E-03

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GML . GAT -	
PROSRAM -	
NC I L WI T M I L	
SWATH	
MAVE RESISTANCE CALCULATIONS FOR SWATH OPTIMIZATION PROSRAM - GML.641 - AUGUST 1980	CHARACTERISTICS
WAVE RESISTANCE	STRUT GEOMETRIC CHARACTERISTICS

						RATIO .				.255E-n3			0003	.097E-03 .597E-04 2968481F-04		11041015.05
						BONY V-L RATIO				CW1 TOT =			CW2 ToT =	CW TOTA:		DE TOTAL
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TSMAX = HS/LS =		AX H 1/1 B B		ぎんしゃ		STRUT V-L RATIO =	-1 0 [-	.<468083E.04	.5416044E.03	<140786E+U4 431E-03	.6620352E+03	.1731256E+U1	1448918E+U4 292E-U3		. 4104764E 04 . 627E -03 . 4937138E+04 1.397E-03	
HS = 9.814.14 FRUDS = .23644	ISTICS	HB = 16.646217 FROUDH = .216256	FEET = 75.226 B/_8	N FEET # 15.233 STRJH OFSEHZLH	WETTED SURFACES ARE FOR A DEMI-JULL	20.254 FPS. 12.000 (MOTS. \$	RESISTANCES IN POUNDS ARE FOR A DEMIJULL	STRUT WAVE DRAG IN POJNDS = RIS/(RHO/20W SJRF*V**2) =	BOOY MAVE DRAG IN POUNUS = REAL(RHO/2*WTSJRF*V**2) =	STRUT-BODY WAVE DRAG IN L3S = 2154/(RH /2**15URF*V**2) =	STRUT INTERFERENCE DR G LBS = 4125/RM /2***ISURF***2: =	830Y INTEAFERENCE DAAG L3S = 3129/(RH /2**TSUAF***2) =	STRUT-800. INTF?NCE D4.G _B = 21253/(R40/2*MTSUHF*V**2) =		STRUT FRICTION DA.G IN LAS = SFS.(FMP.)ZMFTVS-C) = 900Y FRICTION JAAG IN LBS = 4FB.(RMO/ZeMTSJRFeVeez) =	
XLS 227,09-260 GAMADS 8,90-918	BODY GENETRIC CHARACTEMISTICS	XLB 272.62326 GAMAOB 10.691374	SEPARA-ION DISTANCE IN FEET	STAUT CL FROM HODY CL IN FEET	WETTE	SMIP SPEED IS 20.25	RESIS									

.1401034E+05 2.819E-03 515.929

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RF TOTAL = CF TOTAL =

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WAVE RESISTANCE PALCULATIONS FOR SWATH OPTIMIZATION PROGRAM - GML.GM! - AUGUST 1940	CHARACTE-215TICS
WAVE RESISTANCE	STRUT GEOMETHIC CHARACTERISTIC

271-21-24-44-1-24-1-24-1-25-1-25-1-25-1-25	116171717171717171717171717171717171717	ָ ֖֖֖֡֞֝												
KIS 227.0	227.03 260 6.541123	4S =	11 H	9.814-14		TSMAX #	н н	12.485406	.485406					
HODY GEMMETRIC CMARACTEMISTICS	C-A 34CTE- ISTIC	Ş.												
XIH 272,62728	~	#8 # F 40U05 #		16.040617		AX =		146.629250 .061059	.629250 .061059					
SEMARA: ION DISTANCE IN FEET	TANCE IN FEET	11	75.426	8.78		#	٧.	٠٤٦٥.						
STHUT IL FROM MIDY CL IN TEET	ANDY CL IN FEE	n -	15 233		STRUT OFSETZE	7.		18	•050					
	WETTED SUR	FACES ,	ARE F	METTED SURFACES ARE FOR A DEMINAULE	7									
SMIP SPEE IS	S 23.629 #PS. 14.000 (NOTS.	. 14.	000	<no15.< td=""><td>ST</td><td>STRUT V-L RATIO =</td><td>RATI</td><td>II C</td><td>626.</td><td>06</td><td>></td><td>ž 7</td><td>800Y V-L MATIO =</td><td>4</td></no15.<>	ST	STRUT V-L RATIO =	RATI	II C	626.	06	>	ž 7	800Y V-L MATIO =	4
	4ESISTANCE	à NI S	SCNDO	FSISTANCES IN POUNDS ARE FOR A DEMI-HULL	EM] - 4	<u>ار</u>								
	STRU 31S	T #AVE	0416	STRUT MAVE DWIG IN POUNDS 31S/(RMO/ZewTSJAFeveez)	. "	.311A338E+04	388E - 04 . 46 lt 03	, ¬ ¬						
	H3DY 414	4AVE 1	URAS Zewts	HODY MAVE URAS IN POU US RIM/(RMO/ZewISJRF+Ve+2)	H H	.1314265E+0+	265E - 04 . 194E - 03	٠,						
	STPU 415	T-800v	WAVE	STHUT-BODY WAVE DAIG IN L9S		2 92432E+14 309E-U	2432E+14 309E-03	* ⁷		7#1 TOT =	H F-		•34rE=03	
	STRU 412	IT INTEL	¥FЕНЕ /2•▲Т	STRUT INTERFERENCE DW 6 L3S 2125/(RM /2**TSUAF*V**2)	H H	.1869887E+03	887E+113 •02-1E-113	.e. o						
	Y009	INTERI	FEREN	BODY INTERFERENCE DRAG L3S	, 4	.3169488E+ 2	488E - 2	20.3						
	STRU 412	JT -80D !S 37 (R 1)	IN F 0/2*#	STRUT-BOD IN FANCE D4.6 _B 41254/(R4U/2*MTSUMF***2)	. "	4~6R674E.03	8674E.113	2 0		CWZ TUT	# *	•	037E-03	
	α in	UT FRIC	CTION	STRUT FRICTION DR-6 IN Las #	*	.5400577E . U.	177E.	á	0 0₹	CW TOTAL CB AR	H H H	Ŋ,	.309E-03 .809E-03 .5473931E .04	

.2027043E.05 2.997E-03 870.862

F : 3

.1.79650E .05 2.188E-03

RF TOTAL ...

.5499577E.U. .613E-0.3 .-29692E.U* 1.374E-0.3

STRUT FRICTION DM-G IN L3S R
4FS/(RMO/2eMTS)RF4Vee2 R
30DY FRICTION DRAG IN LBS R
2FH/(RMO/2eMTS)RF4Vee2) R

UGUST 1980	.485406 .043217		.061950 .061059		•056		1.062 BONY V-L RATIO =				CW1 TOT = .186E-n3		
POSRAM - GML, GMT - A	TSMAX # 12.485406 HS/LS # .043217		Ax = 146.829250 H-/LB = .061059	× .270	FSET/LA E		STRUT V-L RATIO =	11 4ULL	4194254E .04 = .475E-03	2863267E.04 . 324E-03	541A042E+U4 613E-03	178875E+U2 005E-U3	. 8147801E+U3 . 0946+U3
WAVE MESISTANCE CALCOLATIONS FOR SWATH OPTIMILATION PROSRAM - GML,641 - AUGUST 1980 Strut G-ometric Characteristics	XLS 227,09-250 HS = 9.614~14 GAMAOS 5.009579 F30UDS = .315925	BODY GE -METRIC CHARACTERISTICS	XLB 272,621260 HB = 16,646217 6A4±08 6.01 898 F30UD = .288341	SEPARA ION DISTANCE IN FEET # 75.226 67.6	STAUT (L FROM ANDY CL IN FEET # 15 233 STRUT OFSET/LA	WEITED SURFACES ARE FOR A DEMIJULE	SHIP SPEED IS 27.005 FPS. 10.000 KNUTS.	HESISTANCES IN DOUNDS ARE FOH A DEMI HULL	STRUT WAVE DR.G IN POJNDS = AIS/(RMO/20mTSJRF0V002) =	BODY WAVE DRAG IN POUNUS = 318/(RHO/20MISJRF0V002) =	STRUT-BODY WAVE DAMG IN L35 = 4154/(RH:/ZewTSURFeVee2) =	STRUT INTERFERENCE DN-G L3S = 412S/(RH·/2**TSURF*V**2) =	BODY INTERFERENCE DRAG LAS = AIZ4/(RM /20m15URF0V002) =

969

.2079361E+05 3.033E-03 1315.554

F 0 1

.1906999E .05 2.150E-03

RF TOTAL = CF TOTAL =

.7086923E · 04 .H02E - 03 .1198307E • 03 1.356E - 03

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STRUT FRICTION DAME IN LBS
AFS/(RMO/2*MISJRF*V**)
BODY FRICTION DAME IN LBS
AFS/(RMO/2*MISJRF*V***)

.374E-03 .874E-03

CW TOTAL ...

.189E-03

CW2 TOT =

.4438478E+03

STRUT-BOD: IN:FRNCE DAMG _B = RI253/ (RHU/2*#FSURFev**2) =

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AUGUST	TSMAX = 12.485406 MS/LS = .043217
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OR SMATH OPTIMIZATION PHOSRAM - GML,GMT - AUGUST 1980	9.814414
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ATIONS FOR TERISTICS:	HS = F 40UDS =
STRUT GEOMETRIC CHARACTERISTICS	227.095260 3.95A186
STRUT GEO	XI.S GAMAOS -

					BODY V-L RATIO *
400	Ax = 146.629250 HH/LB = .061059	92	• • 056		STRUT V-L RATIO = 1.19.
,	* *	.276			RATI
	AX HH/LB	•	STRJE OFSEEZLH	_	STRUT V-L
	HB = 16.646217 OUDS = .324384	8.78		WETTED SURFACES ARE FOR A DENI	4015.
		* 75.226	15.23	RE FJ	000
	HB #	*	*	ACES A	30.380 FPS. 17.000 (NOTS.
STICS	ia.	EET	FEET	SURF	rps.
ICTEHI	A.	¥ ₹	CL 13	ETTED	10.380
CHAN	272.62726" 4.751722	STANCE	RODY	,	
ETRIC	272.	IQ NO	FROM		PEED
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		C#1 TOT =			CWZ TOT =	CW TOTAL #
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. *	. "	. "		. "		
STRUT WAVE DRUG IN POJNDS 125/RHD/20WISJRFevee2)	BODY #AVE DRAG IN POUNDS 418/(RMO/20MTSJRF+V++2)	STRUT-BODY WAVE DRIG IN L3S : 21SH/(RHJ/2*WTSURF*V**2)	STRUT INTERFERENCE DR.G L3S : 2125/RM /2*#TSURF*V**2)	BODY INTENFERENCE DRAG L3S : 2129/(RHJ/Z*#ISURF#V**2)	STRUT-BOD INTFANCE DANG _B = 41253/(R40/2-#TSURFevee2)	STRUT FRICTION DRAG IN L9S 4FS/(RHO/2eMTSJRFevee2) 90DY FRICTION DRAG IN LBS 4FB/(RHO/2eWTSJRFevee2)

.3444525F.05 3.081E-03 1902 655

CT

.2385650E.05 2.134E 03

RF TOTAL = CF TOTAL =

WAVE RESISTANCE CALCULATIONS FOR SWATH OPTIMIZATION PROGRAM - GML.647 - AUGUST 1980 Strut Geometric Characteristics

							BO Y V-L HATTO = 1.211				CW1 Tof a			CW2 TUT 156E=n3	CW TOT L .0236:03	PF TUTAL # .2915094E.05 CF TOTAL # 2.112E-13	ě.	EH 2 2247,592
	.434-14 TSMAX # 12.485406		.366217 AX = 145.529250 .360427 H=/LB = .061059	.76	STRUT OFSETZLM * .056		STRUT V-L RATIO = 1.32	JH A DEMI-4ULL	JUNDS = ,7.87906E.04 .*2) = ,542E-03	NUS = .4461345E-04	IN L3S = -,4473140E+04	G L3S = ,3/224]4E.04 **2, = ,2/33E-03	\$ L3S = .8161470E.03	6 _b =h234378E.u4 V=e2) =452E-u3	3S = 1083070E.0. 785E-0.3 5 = .1832024E.0.5			
7.4.7.1.1.1.1.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2	XLS 227.09-260 HS = 9.414-14 GAMADS 3.20-130 FROUDS = .394-UT	BODY GE, WETRIC CHARACTEMISTICS	XLH 272.627260 HB = 16.046217 GAMAOB 3.842895 F30UD7 = .360427	SEPARATION DISTANCE IN FEET # 75.226	STHUT CL FROM HODY CL IN FEET # 15 233	METTED SURFACES ARE FOR A DEMINAULL	SHIP SPEED IS 33.756 FPS. 20.000 KNOTS.	RESISTANCES IN POUNDS ARE FUR A DEMIHULL	STRUT MAVE DRAG IN POUNDS RIS/(RHO/Z#WTSJRF#V##2)	BODY MAVE DRAG IN POUNDS ?IH/(AHO/2*WTSJRF*V**2)	STAUT-BODY WAVE DAMG IN L3S 21S-/(RHJ/204FBU402)	STRUT INTERFERENCE DH G L RIZS/(RH /2º#TSUAFev*+2.	BOOY INTE-FERENCE DALG L 2127/(RM /2**ISU4F*V**2)	STAUT-BODY INTERNCE DG _B 31253/(RAU/Z*MISUMF*V**Z)	STRUT FRICTION DA.G IN L3S AFS/(RHO/C*WISJRF*V**2 300Y FRICTION JRAG IV LBS AFH/(RHO/C*WISJRF*V**2)			

WAVE RESISTANCE CA_CULATIONS FOR SWATH OPTIMIZATION PROGRAM - GML.G4T - AUGUST 19AD STAUT 6:OMETRIC CHARACTERISTICS

					-L RATIO = 1.332				•622E-n3			.364E.03	1.00eE-03 1.861E-03 .310799E.05	34948865.05 2.0925-03
					80nY V-L				CW1 TOT #			C#2 TOT =	CW TOTAL B Co aR	RF TOTAL = CF TOTAL =
12.485406	146.629250	ø	950° ≥		= 1.460		n	ņ	. ~	. •	. ~	. m	ٽ	ក្សក្
TSMAX # 1	Ax = 14	a .276			STRUT V-L RATIO	101.1	.1314653E+05 .787e-03	.8128567E+04	1088232E+05 652E-03	.5694433E.04 .341E-03	.3911885E + 04 .234E - U3	3196318E.04 191E-03	.1298356E - 05	.777E-03 .2196530E-05 1.315E-03
9.614-14 .434-197	16.046617	8~_8 8/_8	233 STRJT OFSETZLH	FOR A DEMINAULL	22.000 (NOTS. S	RESISTANCES IN POUNDS ARE FOR A DEMI JULL	SIN POUNUS = SURFEVEEZ) =	IN POUNDS =	E DRAG IN L3S = TSURFOV#02) =	ENCE DRAG LBS = ISURFOVER2, =	NCE DRAG L3S = ISURF evene) =	FRACE DAMG _B = #ISUNFOVOR;) =		SJRFeveez) # DRAG IN LBS # SJRFeveez) #
HS # F40U05 #	1571CS HB = F40UDc =	FEST = 75.226	N FEET = 15.233	WETTED SURFACES ARE FOR	FPS.	FANCES IN POUND	STRUT MAVE DRAG IN POUNDS 21S/(RHO/2+WISJRF+V++2)	BODY JAVE DRAG IN POUNUS 2187 (RMO/20MISJRF*V**2)	STRUT-BODY MAVE DRAG IN L3S	STRUT INTERFERENCE DR.G LBS 3125/(RH /2+#TSURF+V++2.	BODY INTEAFERENCE DRAG L A124/(RHJ/204TSUAF0V002)	STRUT-BODY INTFANCE DAMG	STRUT FRICTION DWAG IN LAS	4FS/(RMO/20MISJRF0V002) 90DY FRICTION DRAG IN LBS 4FA/(RMO/20MISJRF0V002)
	METRIC CHARACTEMISTICS 272,624260 3,180905 F	SEPARA ION DISTANCE IN FEET	FROM HODY CL IN FEET	BITE	SHIP SPEED IS 37.132	RESISI								
KI S GAMAOS	8007 GECMETRIC XLH 272.6 GAMAOB 3.1	SEPARA	STRUT CL FROM		SHIP									

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HAVE RESISTANCE CA.CLEATIONS FOR SHATH OPTIMICATION PROSRAM - GML.GMT - AUGUST 1980	STRUT GLOMETHIC CHARACTERISTICS

							8007 V-L RATIO # 1.454				1.2536-03			.8816.03	2,134E-03 2,442E-03 .5847572E-05	.4124629E.05 2.074E-03	.9972201E-05 5.017E-53 7344.472
											CW1 TUT =			C#2 TOT =	CW TOTAL = CR RR	RF TOTAL **	# # #
	TSMAX # 12.485406 HS/LS # .043217		AK # 146.629250 H4/L8 # .061059	a15. a	1/L" = .056		STRUT V-L RATIO = 1.593	חרי	.1811587E.05	.1304392E+05 .656E-03	6247330E.04 314E-03	.5-97737E-04	.7014915E+04 .353E-03	.4492953E+ 4 .226E-U3		.1532164E.03 .771E.03 .259260E.03 1.304E.03	
ACTERISTICS	HS 9.614.14	CTEMISTICS	HB = 16,646617 FROUDS = .432512	IN FEET & 75.226 81.8	CL IN FEET # 15 233 STRUT OFSETZL	WETTED SURFACES APE FOR A DEMI-HULL	*0.547 FPS. 2**£03 KNOTS. STI	RESISTANCES IN POUNUS ARE FOR A DEMI HULI	STRUT MANE DRUG IN POUNDS * 21S/(NHO/20HTSJAF6V002) #	HODY MAVE URAN IN POUNUS H	STHUT-BODY WAVE DRAG IN LBS # 4154/(RH:/Ze-t3URFeVe=2) #	STHUT INTERFERENCE DR.6 L3S = 3125/RH /20475URF0V*02) =	BODY INTERFERENCE DRAG LBS = 3124/(RH, /2**T5UMF*V**2) =	STHUT-BOD! INTFANCE D4.6 _B = a1253/(R4U/20#TSUHFevee2) =		STRUT FRICTION DAMG IN L3S # 4FS/RHOLZEWISJRFWEWED) # 30DY FRICTION DRAG IM LBS # 4FH/(AHO/ZEWISJRFEWEMED) #	
STRUT GEOMETHIC CHARACTERISTICS	XLS 227.09-260 GAMAOS 2.22.479	HODY GEUMETRIC CHARACTEMISTICS	K: P 272.027260 GAMAOU 2.677843	SEMARA TON DISTANCE IN FEET	STHUT C FROM ADDY CL IN FEET	HJ	SHIP SPEE, 15	ni œ									

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						BONY V-L RATIO =	
TSMAX # 12 485406		AX = 146.629250 √/L8 = .061059	.276	950. =		.0 = 1.72	
# # × vo			₩			L RATI	
TSHA 1976		AX H 1/18 H		STRUT OFSETZLE	-4	STRUT V-L RATIO =	170F1H2
9.814.14		HB = 16.046217 0UD0 = .468525	8-18		WETTED SURFACES ARE FOR A DEMIHULL	.510	RESISTANCES IN POUNDS ARE FOR A DEMI-HULI
		94 "	= 75.226	15.233	RE FJA	0000	V SONO
#S = F 40005 =		HB ≡ F30U0n ≡		w	ACES A	2,	Od NI
L	C-A-ACTE-ISTICS	Ü.	IN FEET	4707 CL IN FEET # 15.233	TTED SURF	43.883 FPS. 20.000 (NOTS.	SISTANCES
9-260	TA WAC	2356-	LANCE	1707 CL			œ.
227.6	*ETRIC	272.62326° 2.277452	SEPARA: ION DISTANCE IN FEET	STHUT OL FROM H		SHIP SPEED IS	
XLS GAMAUS	HODY GENAETHIC	RLB	SEPARA	STHUT		GIHS	

		1.676E-03			1.0296-03	2,705E-03 3,431E-03	
		CW1 TOT =			CW2 TOT =	CW TOTAL SCR	
.2166027E+05 .924t-03	.1726300E+05	.174093E.03	982214E. 4 .214t-03	.8473783E.04 .372E-03	.1035771E+05 .444E-03	.17843765.05	.3019593E-03 1.294E-03
STRUT MAVE DRAG IN POUNDS = AIS/(AMO/20MTSJRF*V**2) *	BODY WAVE DRAG IN POUNDS = 21H/(RHO/2*MTSJRF*V**2) =	STRUT-BODY WAVE DAMG I LBS = 41S1/(RHJ/2*WTSURF*V**2) *	STRUT INTERFERENCE DM.G L35 = 4125/(RM /20mTSUMF0V002) =	BODY INTEMFERENCE DRAG LBS = 4123/RM//20MF6Vee2) =	STRUT-BOD: INTFANCE DANG B = 31253/(R40/20mISURFevee)	STRUT FRICTION DR.G IN L35 #	RESTRUCTION DRAG IN LUS REFIX (RHOTZENIS BREEZ)

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RF TOTAL ...

JOB CONTROL FOR SWATHO

The Fortran code file of SWATHO is stored on a DTNSRDC Computer Center disk pack. To access the disk pack and make use of the program the following directions should be followed.

One must first set up an account with the DTNSRDC Computer Center and acquire a user ID. A Computer Center Reference Manual will provide the user with a further explanation of the Job Setup Control cards.

1. To get the program from the disk pack to a permanent file:

JOB CARD
CHARGE CARD
PAUSE. JOB REQUIRES DV4865.
MOUNT, VSN=DV4865, SN=CHRELIB.
REQUEST, LOG, *PF.
ATTACH, OLDPL, OPTSWATH, ID=CHRE, SN=CHRELIB.
UPDATE(P,F)
FTN(I=COMPILE, R=3)
CATALOG, LOG, USER'S NAME OF FILE, ID=USER ID
7/8/9
6/7/8/9

2. To run the program which is stored on the user's permanent file:

JOB CARD
CHARGE CARD
ATTACH, USER'S NAME OF FILE, ID=USER ID.
FIRST SEVEN LETTERS OF NAME OF FILE (PL=50000)
7/8/9
INPUT CARDS
6/7/8/9

The user's permanent file which is set up with the catalog card in Part 1 will be purged after 30 days of inactivity.

MAINTENANCE MANUAL

This part of the report contains a description of the main program, SWATHO and the subroutines directly called by it as well as descriptions of how other important subroutines fit together. Also included in this manual are descriptions of the common blocks, functions and subroutines and a functional listing of the functions and subroutines.

INTRODUCTION

The main program, SWATHO, first calls subroutine INPUT. Subroutine INPUT reads the input cards prepared by the user. The strut geometric characteristics, waterplane area and moment coefficients as well as the height of the CG above the baseline are read. Next, the body geometric characteristics, separation of hull centerlines, placement of the strut centerline in relation to the body centerline and the body prismatic and moment coefficients are read. Following that the S array of scaling factors, the optimization speed and limiting values are read.

After returning from INPUT the program calls PRAXIS, the optimization routine, which is described in detail in the following pages of this report.

Upon the completion of the optimization by PRAXIS, the subroutine SCALE is called which sets up an array of current working values of X, each member of the array corresponding to one of the ship geometric characteristics. The X array is calculated by multiplying S by XI where the S's are the scale factors and the XI's are the initial values of the ship geometry. Both S and XI are read by subroutine INPUT. Initially the members of the S array all have values of 1 and then, on subsequent runs, the S's correspond to the lowest value of EHP so far encountered by the user in the output from the previous run of the program. All of the initial variables are left unchanged.

Subroutine CHEB, called by the main program next, calculates the Chebychev coefficients for the ship as well as the wetted surface of the body and strut.* It also calls another subroutine, PCHEB, which plots the strut waterplane and body sectional area curves.

LISTEX is then called and values ϵ ? ship geometry, etc. are listed to display the proximity of the solution to the extreme limits.

^{*} See Appendix C for a discussion of the relationship between the Chebychev coefficients and the hullform coefficients.

Function EHP is lastly called at which time final design resistance and power is output. Whether the program gets this far is contingent upon how much time the computer is given for the computations.

Subroutine PRAXIS accomplishes the optimization of the objective function, EHP. PRAXIS finds the minimum of a function using the principal axis method as implemented by Brent. The arguments of the subroutine are TO, HO, N, IPRIN, X, F, and FMIN. PRAXIS attempts to find this minimum such that if XO is the true local minimum near X, then NORM (X-XO)=TO + SQUARE ROOT (MACHEP) *NORM (X), where MACHEP is the machine precision, the smallest number such that 1 + MACHEP>1, and TO is the tolerance.

The values of TO, HO, N and IPRIN are set in the data statement in SWATHO. The value of MACHEP should be 2**-47 (about 7.105 E-15) on the CDC 6000 series for single precision arithmetic or 16**-13 (about 2.23 E-16) for double precision on the IBM 370 system. HO is the maximum step size, and should be set to about the maximum distance from the initial guess to the minimum. This value of HO affects the initial rate of convergence, in the current program HO = .25. N is the number of variables upon which the function depends, in this case N = 12. N must be at least 2. IPRIN controls the printing of intermediate results of the optimization.

X is an array which contains a guess of a point of minimum and an estimated point of minimum, on entry and return, respectively. F(X,N) is the function to be minimized, in our case it is PEHP. F needs to be a real function declared EXTERNAL in the calling program. FMIN is set to the minimum value of F that is found.

The approximating quadratic form is

$$Q(X') = F(X,N) + \frac{1}{2} * (X' - X) TRANSPOSE *A* (X' - X).$$

Here A is INVERSE (V-Transpose) *D* INVERSE (V) where V(*,*) is the matrix of search directions and D(*) is the array of second differences.

PRAXIS in operation proceeds as follows. PRAXIS calls PRMIN which minimizes the function along first direction V(*,J). PRMIN then calls function PRLIN which is a function of one real variable, which it minimizes. Function RANDUM is called and a random number is returned to PRAXIS to avoid resolution valleys. PRMIN continues to minimize the PRLIN function in different directions. PRQUAD is then

called and this subroutine tries a quadratic extrapolation in case minimization is being done in a curved valley. PRAXIS then calls PRFIT to find principal values and directions of the quadratic forms and PRSORT is called to sort the eigenvalues and eigenvectors. PRPRIN prints out the scaling factor on the first page of the output as is described in the output section of the USER's manual and VCPRINT prints the second difference array.

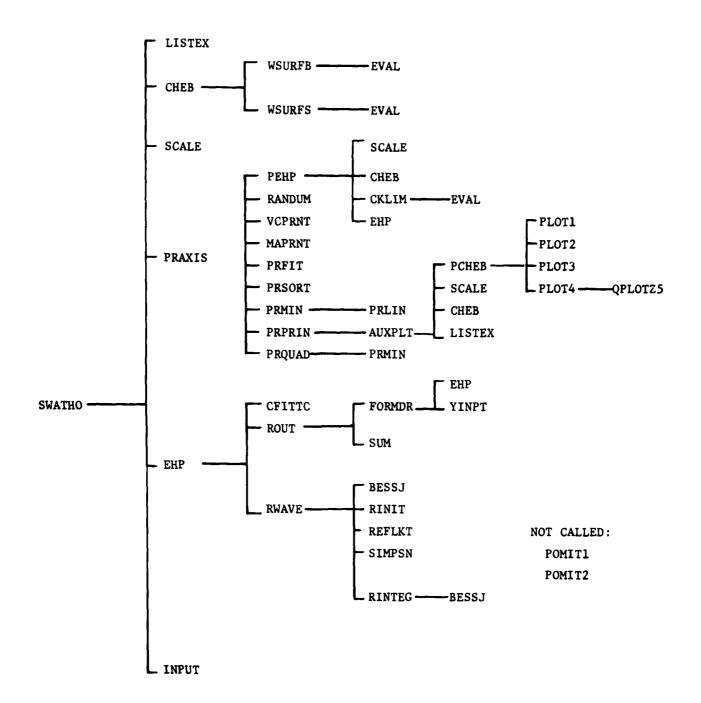
The optimization procedure is performed on the function PEHP. PEHP gets its value from the EHP calculated by the function by that name and a penalty P, so that PEHP = $(1 + P)^*$ EHP. P is determined from an array calculated in subroutine CKLIM. CKLIM uses the constraints determined by subroutine INPUT to check the current ship values for violation of these constraints. The checking procedure is quantified with an array GG of violation coefficients, in the following way. The current value of ship displacement is multiplied by a member of a scaling array, ALPHA, given in a data statement in CKLIM. This results in a violation coefficient GG(1) for displacement. Nineteen of these coefficients are calculated and this array of GG(I)'s is used in function PEHP. Since GG(I)>0 means that the constraint is violated whereas GG(I)<0 means that the constraint is satisfied, the function PEHP totals up all of the GG(I)'s greater than zero and this becomes the value of P, used to calculate PEHP. For further details on penalty functions see Appendix A.

Other important subroutines have not been mentioned yet. Subroutine RWAVE, called by Function EHP, computes the auxiliary wave resistance functions T and W as explained in Appendix B, using the Chebychev expansions calculated in subroutine CHEB, as discussed in Appendix C. Subroutine RINTEG integrates these auxiliary functions with subroutine INIT initializing the arrays. The symmetric matrices of T and W are reflected by subroutine REFLKT. Using the Chebychev coefficients, subroutine WSURFB determines the wetted surface of the body of revolution and WSURFS determines the wetted surface of the strut.

Subroutine ROUT is called by function EHP, it computes values of Froude number and speed-length ratio as well as wave resistance coefficients and frictional drag coefficients and returns the value of EHP to function EHP. It also prints the intermediate and final design values. Which one of these is printed is determined by the value of KEY. If KEY = 0 no output is printed, instead only the computations in ROUT are performed. If KEY = 1 the optimization is complete and just the final design is printed. If KEY = 2 when ROUT gets to the output section,

the intermediate results are printed. The different values of KEY are set in the subroutines which call FUNCTION EHP which in turn calls ROUT; KEY is set to 0 in PEHP, 1 in SWATHO and 2 in AUXPLOT.

Figure 2 - TREE DIAGRAM OF SWATH PROGRAM



FUNCTIONAL LISTING OF FUNCTIONS AND SUBROUTINES

Subroutine Name	Description
AUXPLT	Prints titles and labels for body sectional area and waterplane curves
BESSJ	Evaluates Bessel function
CHEB	Determines Chebychev coefficients for body and strut
CKLIM	Checks for violation of geometric constraints
INPUT	Reads in initial values of ship geometry and constraints
LISTEX	Lists various values of the design to see how close to the limits the solution lies
MAPRNT	Prints columns of matrix
РСНЕВ	Plots body sectional area curve and water- plane outline curve
PLOT1	Sets up spacing and determines values of the axes for printer plot
PLOT2	Establishes formula for computing location in the image region where the point will be plotted, in the printer plot
PLOT3	Assigns an alpha character to each point to be plotted on the printer plot
PLOT4	Prints image of completed graph on the printer
POMIT1	Causes certain grid lines on printer plot to be deleted
POMIT2	Causes values at grid lines to be deleted
PRAXIS	Finds the minimum of the objective function using the principal axis method
PRFIT	Fits points to a quadratic curve
PRMIN	Minimizes the objective function, F, in one dimension
PRPRIN	Prints intermediate results and calls the routine which does the printer plotting

FUNCTIONAL LISTING OF FUNCTIONS AND SUBROUTINES (CONT)

ong a quadratic
matrix to find the
ation to label the
liary wave resistance
wave resistance
f the auxiliary
stance s, and also and final design
ve resistance
of ship geometry
iers for numerical rule
of wave resistance on involving the additional the auxiliary matrices
prints the scale factors, value of the the
of a body of ychev coefficients
of strut whose s given by
of of of

FUNCTIONAL LISTING OF FUNCTIONS AND SUBROUTINES (CONT)

Function Name	Description				
CFITTC	Determines frictional resistance coefficient from ITTC line				
ЕНР	Calculates body constants and frictional drag coefficients				
EVAL	Evaluates the Chebychev series				
FORMDR	Evaluates form drag coefficients				
PEHP	The objective function to be minimized by PRAXIS				
PRLIN	Function of one variable minimized by PRAXIS				
RANDUM	Returns a random number which creates random steps during the optimization				
YINTP	Interpolates through a set of discrete data				

COMMON BLOCK DESCRIPTIONS

COMMON/COEFS

PURPOSE: COEFS stores the values of the Chebychev coefficients for the strut and body

NAME	TYPE	LENGTH	DEFINITION
ASM	R	(3)	Coefficients of Chebychev Sine Series for strut
BSM	R	(3)	Coefficients for Chebychev Cosine Series for strut
ABM	R	(3)	Coefficients of Chebychev Sine Series for body
ВВМ	R	(3)	Coefficients of Chebychev Cosine Series for body
MMAX	1		Maximum order of Chebychev Series

COMMON/EXTLIM

 $\begin{array}{ll} \text{PURPOSE:} & \text{EXTLIM defines and stores physical constants and geometric} \\ & \text{parameters} \end{array}$

NAME	TYPE	DEFINITION
DISPMN	R	Minimum ship displacement in long tons
DFAC	R	Minimum draft/body diameter ratio
DRFTMX	R	Maximum draft of ship
WMAX	R	Maximum width of ship
XFWD	R	Minimum distance from body L.E. to strut L.E.
XAFT	R	Minimum distance from body T.E. to strut T.E.
MINGMT	R	Minimum GMT (transverse GM)
MINGML	R	Minimum GML (longitudinal GM)
LODMAX	R	Maximum length over diameter ratio
LODMIN	R	Minimum length over diameter ratio
TSMIN	R	Minimum strut thickness at 50 percent chord
KG	R	Height of center of gravity of ship above baseline

COMMON/INITL

PURPOSE: Store initial values of ship geometry

NAME	TYPE	DEFINITION
XLSI	R	Length of strut in ft.
нѕі	R	Draft of strut in ft.
TSMAXI	R	Max. strut thickness in ft.
CWPI	R	Waterplane area coefficient
CLCFI	R	Waterplane moment coefficient
CIYYI	R	Waterplane inertia coefficient
CLCBI	R	Body moment coefficient
XLBI	R	Length of body in ft.
BDIAI	R	Body diameter in ft. at XLB/2
BSI	R	Separation of hull centerlines
CPI	R	Body prismatic coefficient
CSTRTI	R	Dist. of strut CL fwd of body CL
VBI	R	Initial displaced volume of the body in cubic ft.
VSI	R	Initial displaced volume of the strut in cubic ft
KBI	R	Initial height of center of buoyancy above the baseline
BMLI	R	Initial longitudinal metacentric height
IYYI	R	Waterplane moment of inertia

COMMON/NOCHG

PURPOSE: Checks whether certain ship geometry variables change or not.

NAME	TYPE	LENGTH	DEFINITION
LL	I	6	LL(I) thru LL(6) represents positions in WORKVL of XLS, BS, HS, XLB, BDIA CSTRT. In the data statement in SWATHO they are assigned values; LL(1) = 1, LL(2) = 10, LL(3) = 2, LL(4) = 8, LL(5) = 9, LL(6) = 1
SS	R	6	Holds array of scale factors corresponding to the six ship geometry variables in the LL array
SAME	L		Variable which indicates whether scaling variables have changed or not.

COMMON/OMEGA

PURPOSE: OMEGA stores the values of variables and constants for evaluation of auxiliary wave resistance functions and ship resistance coefficients

NAME	TYPE	DEFINITION
VMFPS	R	Speed of ship (feet per second)
GAMAOS	R	γ_{os}
GAMAOB	R	γ_{ob}
GOSQ	R	$(\gamma_{os})^2$
HSOLS	R	Ratio of draft to length of strut
HBOLB	R	Ratio of draft to length of body
WETS	R	Wetted surface area of strut (ft ²)
WETB	R	Wetted surface area of body (ft ²)
WTSURF	R	Total wetted surface area (ft ²)
SEP	R	$2b (\gamma_{os}L_s)$
PHIS	R	2(h _s)
		L _s Y _{os}
PHIB	R	2(h _b)
		LYob
RATIOL	R	γ_{ob}/γ_{os}
CFS	R	Frictional drag coefficient of strut
CFB	R	Frictional drag coefficient of body

COMMON/PHYSCO

PURPOSE: PHYSCO stores physical constants

NAME	TYPE	DEFINITION
RHO	R	Density of water
GNU	R	Kinematic viscosity of water
G	R	Acceleration due to gravity
PI	R	Ratio of circumference to diameter of a circle
DELCF	R	Correlation allowance
KTFPS	R	Conversion factor for changing ft/sec into knots

COMMON/PRCOMM

PURPOSE: Used by optimization subroutines to transmit values amongst themselves

NAME	TYPE	DEFINITION
FX	R	Function to be minimized
LDT	R	Used in step size computations
DMIN	R	The square of the machine precision
NF	I	Function evaluation counter
NL	ı	One dimensional search counter

COMMON/PSI

PURPOSE: PSI stores the values of the variables and constants used in integration routines

NAME	TYPE	DEFINITION
NPTSZ	I	Number of integration steps from γ_{os} to γ_{os}^{+1}
PTSAF	R	Scaling factor of step size in integrating from α to α smax
EXPN	R	Empirical constant for integration to stop
NALMAX	I	Maximum number of integration steps from γ +1 to α max
NAL	I	Counter of integration steps
ALFA	R	Integration variable (α)
ALSMAX	R	Maximum of α for integration

Comments: Values of NPTSZ, PTSAF, EXPN, NALMAX and ALSMAX are assigned in data statement in SWATHO

COMMON/Q

PURPOSE: Used by optimization subroutines to transmit function representations amongst themselves

NAME	TYPE	DEFINITION
v	R	Matrix to define directions for linear search
QØ	R	Variables which define the plane for the quadratic search
Q1	R)	
QA	R	
QB	R	Variables which define a parabolic space curve
QC	R)	
QD Ø	R)	Variables used in QA, QB, QC to define
QD1	R	quadratic curve
QF1	R	Value of function defined by QA, QB, QC

COMMON/SPEEDS

PURPOSE: Stores speeds at which resistance colculations are made

NAME	TYPE	LENGTH	DEFINITION
NSPEED	I		Number of speeds
SPEED	R	20	Speeds in feet per second

COMMON/XRPLOTF

PURPOSE: XRPLOTF stores the values of variables for the plotting routines ${\bf var}$

NAME	TYPE	DEFINITION
XL	R	Value of abscissa at left-most grid line
XH	R	Value of abscissa at right-most grid line
YL	R	Value of ordinate at bottom grid line
YH	R	Value of ordinate at top grid line
ХI	R	Increment of divisions along the abscissa
YI	R	Increment of divisions along the ordinate
YMOV	R	Ordinate index increment number of array GRAF
XMOV	R	Abscissa index increment number of array GRAF

COMMON/XRPLOTG

PURPOSE: XRPLOTG stores the values and characters of variables for the plotting routines

NAME	TYPE	LENGTH	DEFINITION
GRAF	I	(11,204)	Array containing the image to be plotted

COMMON/ XRPLOTQ

PURPOSE: XRPLOTQ stores the constants and characters for the plotting routines ${\bf r}$

NAME	TYPE	DEFINITION
I, II	I	Ordinate scale factor is 10^{I}
J, JJ	I	Number of digits following ordinate decimal point
K, KK	I	Abscissa scale factor is 10 ^K
L, LL	I	Number of digits following abscissa decimal point
A, NHL	I	Integer number of horizontal grid lines
G, NSBH	I	Integer number of spaces beyond each horizontal grid line to next grid line
C, NVL	1	Integer number of vertical grid lines
D, NSBV	I	Integer number of spaces beyond each vertical grid line to next grid line
E, HCHAR		Horizontal grid character
F, VCHAR		Vertical grid character
M, IX, ISX	I	Number of horizontal spaces
N, IY, ISY	I	Number of vertical spaces
v	L	Logical variable = TRUE when maximum and minimum values of the ordinate are determined
н	L	Logical variable = TRUE when maximum and minimum values of the abscissa are determined

COMMON/WORKVL

PURPOSE: WORKVL stores the working values of the ship geometry

NAME	TYPE	DEFINITION
XLS	R	Length of strut in ft.
HS	R	Draft of strut in ft.
TSMAX	R	Max strut thickness in ft.
CWP	R	Waterplane area coefficient
CLCF	R	Waterplane moment coefficient
CIYY	R	Waterplane inertia coefficient
CLCB	R	Body moment coefficient
XLB	R	Length of body in ft.
BDIA	R	Body diameter in ft. at XLB/2
BS	R	Separation of hull centerlines
СР	R	Body prismatic coefficient
CSTRT	R	Dist. of strut CL fwd of body CL
AX	R	Maximum body cross-sectional area in sq. ft.
НВ	R	Body centerline submergence
DISP	R	Displacement of total ship in long tons
AWP	R	Waterplane area of one strut
NLOC	I	Switch indicating presence of second strut
CSTRT2	R	Dist. of strut CL fwd of body CL
OVMFPS	R	Optimization speed in ft/sec

COMMON/WORKVL (CONT)

NAME	TYPE	DEFINITION
КВ	R	Height of center of buoyancy above baseline
BML	R	Longitudinal metacentric height
BMT	R	Transverse metacentric height

SUBROUTINE AND FUNCTION DESCRIPTIONS

NAME:

SUBROUTINE AUXPLT

PURPOSE:

Prints title and values for ship geometry on the body sectional area and waterplane outline curve. Subroutine PCHEB prints the actual curve

CALLING SEQUENCE:

Call AUXPLT (S, N, HP)

ARGUMENTS:

S - Scaling values for each of the variables

N - The number of variables upon which the

function depends

HP - Horsepower

COMMON BLOCKS:

INITL, NOCHG, PHYSCO, SPEEDS

SUBROUTINES CALLED:

SCALE, CHEB, LISTEX, PCHEB, EHP

CALLED BY:

PRPRIN

SUBROUTINE BESSJ

PURPOSE:

SUBROUTINE BESSJ evaluates the Bessel function

from order O to order N

CALLING SEQUENCE:

CALL BESSJ (X, N, VJ)

ARGUMENTS:

X - Argument of the Bessel function
 N - Maximum order of the Bessel function
 VJ - Array holding (N+1) values of the Bessel function of order zero up to N, where

 $VJ(1) = J_o(X)$

•

 $VJ(N+1) = J_N(X)$

COMMON BLOCKS:

NONE

SUBROUTINE CALLED:

NONE

CALLED BY:

RINTEG, RWAVE

FUNCTION CFITTC

PURPOSE:

Function CFITTC determines the frictional resistance coefficient based on the ITTC

correlation line

CALLING SEQUENCE:

C = CFITTC (RN)

ARGUMENT:

RN - Reynolds number at test condition

COMMON BLOCKS:

NONE

SUBROUTINE CALLED:

NONE

CALLED BY:

Function EHP

COMMENTS:

 $C = \frac{0.075}{(\log_{10} (RN) - 2)^2}$

SUBROUTINE CHEB

PURPOSE:

Determines Chebychev coefficients for the ship, and calculates wetted surface of body

and strut

CALLING SEQUENCE:

CALL CHEB (TITLE, KEY)

ARGUMENTS:

TITLE - Label printed on output

KEY - Variable used to skip or include the printed

output.

COMMON BLOCKS:

COEFS, OMEGA, PHYSCO, WORKVL

SUBROUTINES CALLED:

WSURFB, WSURFS

CALLED BY:

SWATHO, PEHP, AUXPLT

SUBROUTINE CKLIM

PURPOSE:

Checks for violation of constraints and scales penalties by the values in the

array ALPHA

CALLING SEQUENCE:

CALL CKLIM (GG, NGG)

ARGUMENTS:

GG - actual amount of violation

NGG - number of values to be checked for

violation

COMMON BLOCKS:

COEFS, EXTLIM, WORKVL

SUBROUTINES CALLED:

EVAL

CALLED BY:

Function PEHP

FUNCTION EHP

PURPOSE:

Calculates body constants and frictional drag coefficients and calls ROUT to

calculate EHP

CALLING SEQUENCE:

EH = EHP (SPEED, TITLE, KEY)

ARGUMENTS:

SPEED - Speed of ship in feet per second

TITLE - Label printed on output

KEY - Variable to skip or include

the printed output

COMMON BLOCKS:

NOCHG, OMEGA, PHYSCO, WORKVL

SUBROUTINES CALLED:

ROUT, RWAVE, CFITTC

CALLED BY:

SWATHO, AUXPLT, PEHP

FUNCTION EVAL

PURPOSE:

Evaluates the Chevychev eries

MAX

 $F(x) = \sum_{X} A(M) *U(M,X) B(M) *V(M,X)$

M=1

CALLING SEQUENCE:

EV = EVAL (X,A,B,MAX)

ARGUMENTS:

X - Arguments of the Chebychev series

A - Coefficients of the Chebychev

Cosine series

B - Coefficients of the Chevychev Sine

series

MAX - Maximum order of the Chebychev series

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

NONE

CALLED BY:

CKLIM, WSURFB, WSURFS

COMMENTS:

 $U_{M}(X) = \cos \{(2M-1)(\theta)\}$

 $V_{M}(X) = SIN 2M\Theta$

 $\theta = \sin^{-1}(X)$

FUNCTION FORMOR

PURPOSE:

Function FORMDR evaluates the form

drag coefficient

CALLING SEQUENCE:

FOR = FORMDR (VL)

ARGUMENT:

VL = Speed-length ratio of ship based

on strut length

COMMON BLOCKS:

NONE

SUBROUTINE CALLED:

FUNCTION YINTP

CALLED BY:

ROUT

COMMENTS:

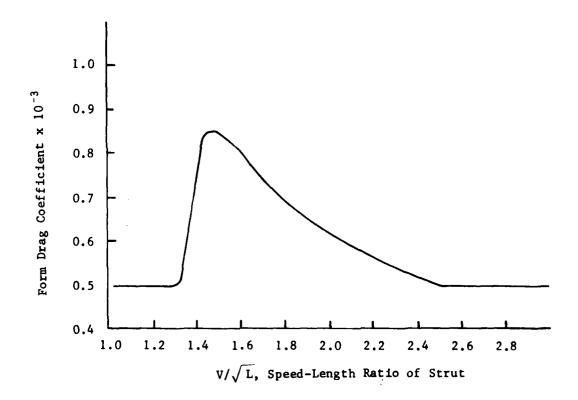


Figure 3. FORM DRAG COEFFICIENT

SUBROUTINE INPUT

PURPOSE:

Reads into common block/INITL/the initial values of ship geometry and constraints, concurrently defining array XI. It then prints the starting point, along with an

echo of the input values

CALLING SEQUENCE:

CALL INPUT (TITLE, S, OPYSPD)

ARGUMENTS:

TITLE - Label printed on output

S - Scale factors

OPTSPD - Optimization speed in knots

COMMON BLOCKS:

EXTLIM, INITL

SUBROUTINES CALLED:

NONE

CALLED BY:

SWATHO

NAME: SUBROUTINE LISTEX

PURPOSE: Lists various constraint values so

that it can be determined how close to the extreme limits this solution

lies

CALLING SEQUENCE: CALL LISTEX

ARGUMENTS: NONE

COMMON BLOCKS: EXTLIM, PHYSCO, WORKVL

SUBROUTINES CALLED: NONE

CALLED BY: SWATHO, AUXPLT

SUBROUTINE MAPRNT

PURPOSE:

Prints the columns of the NxN Optimization matrix V with a heading as specified by OPTION

CALLING SEQUENCE:

CALL MAPRNT (OPTION, V, M, N)

ARGUMENTS:

OPTION = 1 LABELS NEW DIRECTION
2 LABELS PRINCIPAL AXES

V - Logical variable = TRUE
M - The maximum step size
N - Number of variables

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

NONE

CALLED BY:

SUBROUTINE PCHEB

PURPOSE:

Subroutine PCHEB plots by line printer the body sectional area curve and waterplane outline curve from the given Chebychev coefficients

CALLING SEQUENCES:

CALL PCHEB (AS, BS, AB, BB, NN, TITLE)

ARGUMENTS:

AS - Coefficients of Chebychev Sine Series for strut (Symmetric) BS - Coefficients of Chebychev Cosine

Series for strut (Antisymmetric)

AB - Coefficients of Chebychev Sine

AB - Coefficients of Chebychev Sine Series for body (Symmetric)

BB - Coefficients of Chebychev Cosine Series for body (Antisymmetric)

NN - Dimension of AS, BS, AB, BB

TITLE- Array containing the alphanumeric characters of the title of the

project.

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

PLOT1, PLOT2, PLOT3, PLOT4

CALLED BY:

AUXPLT

FUNCTION PEHP

PURPOSE:

PEHP is the objective function to be

minimized by PRAXIS

CALLING SEQUENCE:

PE = PEHP (S,N)

ARGUMENTS:

S - Array of scale factors being optimized N - Maximum order of the Bessel function

COMMON BLOCKS:

OMEGA, WORKVL

SUBROUTINES CALLED:

SCALE, CHEB, CKLIM, EHP

CALLED BY:

PURPOSE:

CALLING SEQUENCE:

ARGUMENTS:

SUBROUTINE PLOT1

Subroutine PLOT1 sets up spacing and determines the values of the axes

CALL PLOTI (NSCALE, A, B, C, D, E, F)

NSCALE - Integer array defined as follows:

NSCALE (1) - I, if printed, values of the ordinate are 10 ** I times the actual values

NSCALE (2) - J, if printed, values of the ordinate are 10 ** J times the actual values

NSCALE (3) - K, if printed, values of the abscissa are 10 ** K times the actual values

NSCALE (4) - L, if printed, values of the abscissa are 10 ** L times the actual values

A - Integer number of horizontal grid lines

B - Integer number of spaces between each horizontal grid line

C - Integer number of vertical grid lines

D - Integer number of spaces between each vertical grid line

E - Horizontal grid character

F - Vertical grid character

COMMON BLOCK:

SUBROUTINE CALLED:

CALLED BY:

XRPLOTQ

NONE

SUBROUTINE PLOT2

PURPOSE:

Subroutine PLOT2 examines the minimum and maximum values of the abscissa and the ordinate and establishes an internal formula for computing location in the image region corresponding to the point to be plotted

CALLING SEQUENCE:

CALL PLOT2 (XMAX, XMIN, YMAX, YMIN, NSCLI)

ARGUMENTS:

XMAX - Value of abscissa at rightmost grid line

XMIN - Value of abscissa at leftmost grid line

YMAX - Value of ordinate at top grid line

YMIN - Value of ordinate at bottom grid line

NSCLI - Logical flag (should be FALSE, if PLOT1 has not been called and standard grid is desired)

COMMON BLOCKS:

XRPLOTF, XRPLOTQ, XRPLOTG

SUBROUTINE CALLED:

NONE

CALLED BY:

SUBROUTINE PLOT3

PURPOSE:

Subroutine PLOT3 assigns an alphacharacter to each point to be plotted

CALLING SEQUENCE:

CALL PLOT3 (PCHAR, X, Y, SDATA, FDATA,

DDATA)

ARGUMENTS:

- Plotting character PCHAR - Array containing the X Х coordinates to be plotted

- Array containing the Y coordinates to be plotted

- Integer position in the arrays SDATA of the first ordered pair to

be plotted

- =1 if each point from SDATA FDATA to DDATA is to be plotted =2 if every other point is to

be plotted

=3 if every third point is to

be plotted

- Integer position in the array DDATA of the last ordered pair to be

plotted

COMMON BLOCKS:

XRPLOTF, XRPLOTG

SUBROUTINE CALLED:

NONE

CALLED BY:

SUBROUTINE PLOT4

PURPOSE:

Subroutine PLOT4 prints the image of the completed graph on the printer, including the values of the abscissa and the ordinate at the grid lines outside the bottom and left edge of the graph

CALLING SEQUENCE:

CALL PLOT4 (MCHAR, NCHAR)

ARGUMENTS:

MCHAR - Single dimension array containing alpha characters to be plotted at the left

of the graph

NCHAR

- Number of valid characters in

MCHAR

COMMON BLOCKS:

XRPLOTF, XRPLOTG, XRPLOTQ

SUBROUTINE CALLED:

QPLOTZ5

CALLED BY:

SUBROUTINE POMIT1

PURPOSE:

To cause certain grid lines on graph to be deleted, depending upon which arguments

are labeled TRUE

CALLING SEQUENCE:

CALL POMITI (T, B, L, R)

ARGUMENTS:

T = top line on graph
B = bottom line on graph
L = left line on graph
R = right line on graph

COMMON BLOCKS:

XRPLOTG, XRPLOTQ

SUBROUTINES CALLED:

NONE

CALLED BY:

Main program

COMMENTS:

This subroutine is part of a plotting package and is included in this program only to keep this package

intact

SUBROUTINE POMIT2

PURPOSE

Causes horizontal and/or vertical values

at grid lines to be deleted

CALLING SEQUENCE:

CALL POMIT2 (ARG)

ARGUMENTS:

ARG

ARG = 1 values of abscissa at

grid lines deleted

ARG = 2 values of ordinate at

lines deleted

ARG = 3 both sets of values

deleted.

COMMON BLOCKS:

XRPLOTQ

SUBROUTINES CALLED:

NONE

CALLED BY:

Main program

COMMENTS:

This subroutine is part of a plotting package and is included in this program

only to keep this package intact

SUBROUTINE PRAXIS

PURPOSE:

Finds the minimum of the function F(X,N) of n variables using the principal axis method. The gradient of the function is not required.

CALLING SEQUENCE:

CALL PRAXIS (TO, HO, N, IPRIN, X, F, FMIN)

ARGUMENTS:

TO - is a tolerance
HO - is the maximum step
size
N - the number of variables
upon which the function
depends (must be at
least two)

IPRIN - controls the printing of intermediate

results

X - contains on entry a guess of the point of minimum, on return the estimated point of minimum

point of minimum the function to be

minimized

FMIN - is set to the minimum

found

COMMON BLOCKS:

PRCOMM, Q

SUBROUTINES CALLED:

PRPRIN, PRMIN, PRQUAD, MAPRNT, PRFIT,

PRSORT, VCPRNT, RANDUM, PEHP

CALLED BY:

SWATHO

COMMENTS:

The approximating quadratic form is:

 $s(X') = F(X,N) + \frac{1}{2} *((X'-X) - Transpose *A* (X-X'))$

X is the best estimate of the minimum A is Inverse (V-Transpose) *D* Inverse (V) V(**) is matrix of search directions

V(**) is matrix of search directions D(*) is array of second differences

SUBROUTINE PRFIT

PURPOSE:

An improved version of MINFIT restricted to $M = N, P = \emptyset$. The singular values of the array AB are returned to Q, and AB is overwritten with the orthogonal matrix V such that U DIAG (Q) = AB.V, where U is another orthogonal matrix.

CALLING SEQUENCE:

CALL PRFIT (M, N, MACHEP, TOL, AB, Q)

ARGUMENTS:

 ${\tt M}$ - is the maximum step size

N - the number of variables upon which the function depends

MACHEP - is the machine precision, the smallest number such that 1+MACHEP>1. MACHEP should be 2**-47 (about 7.105 E-15) for single precision arithmetic on the CDC 6000 system or 16**-13 (about 2.23 D-16) for double precision on the IBM 370 system.

TOL - Tolerance

AB - Array for which singular values are to be determined

Q - Where the original values of the array are to be stored.

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

NONE

CALLED BY:

FUNCTION PRLIN

PURPOSE:

PRLIN is the function of one real variable L that is minimized by the subroutine PRMIN

CALLING SEQUENCE:

PR = PRLIN (N, J, L, F, X, NF)

ARGUMENTS:

N - Number of variable

J - Defines direction for search/ minimization in the V matrix

L - The single variable in the rotated coordinate system upon which the function is assumed to vary

quadratically

F - The function to be minimized

X - Coordinate of the initial point in the quadratic space

NF - The function evaluation counter

COMMON BLOCKS:

Q

SUBROUTINES CALLED:

NONE

CALLED BY:

PRMIN

SUBROUTINE PRMIN

PURPOSE:

Minimizes F from X in the direction V(*,J) unless J is less than 2, when a quadratic search is made in the plane, defined by $Q /\!\!\!/ \!\!\!/ \!\!\!/ , Q /\!\!\!\!/ \!\!\!/ , X$

CALLING SEQUENCE:

CALL PRMIN (N, J, NITS, D2, X1, F1, FK, F, X, T, MACHEP, H)

ARGUMENTS:

- N is the number of variables upon which the function depends.
- J Defines direction for search/minimization in the V matrix.
- NITS Controls the number of times an attempt will be made to halve the interval
 - $\mbox{D2}$ is either zero or an approximation to half $\mbox{\bf F}$
 - X1 is estimate of the distance from X to the minimum along V (*, J)
 - F1 PRLIN (N, J, X1, F, X, NF)
 - FK Logical operator
 - F The function to be minimized
 - X Coordinate of the initial point in the quadratic space
- T Tolerance
- MACHEP Machine precision
 - H Step size

COMMON BLOCKS:

PRCOMM, Q

SUBROUTINES CALLED:

PRLIN

CALLED BY:

PRAXIS, PRQUAD

SUBROUTINE PRPRIN

PURPOSE:

PRPRIN prints intermediate optimization results including the number of function evaluations, function value, and current

coordinates

CALLING SEQUENCE:

CALL PRPRIN (N,X, IMPRIN)

ARGUMENTS:

N - the number of variables upon which the function depends

 \boldsymbol{X} - contains on entry a guess of the point of minimum, on return the estimated point of minimum.

IPRIN - controls the printing of intermediate

results

COMMON BLOCKS:

PRCOMM

SUBROUTINES CALLED:

AUXPLT

CALLED BY:

SUBROUTINE PRQUAD

PURPOSE:

Looks for the minimum of F along a quadratic curve defined by $Q\emptyset$, Q1, X, in case minimization is being done in

a curved valley

CALLING SEQUENCE:

CALL PRQUAD (N, F, X, T, MACHEP, H)

ARGUMENTS:

N - Number of variable

F - The function to be minimizedX - Coordinate of the initial point in the quadratic space

H - Step size

COMMON BLOCKS:

PRCOMM, Q

SUBROUTINES CALLED:

PRMIN

CALLED BY:

SUBROUTINE PRSORT

PURPOSE:

Sorts the elements of D(N) into descending order and moves the corresponding columns of V(N,N)

resulting in eigenvalues and eigenvectors

CALLING SEQUENCE:

CALL PRSORT (M, N, D, V)

ARGUMENTS:

M - The maximum step size
 N - Number of variables
 D - Second difference array
 V - Matrix of search directions

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

NONE

CALLED BY:

SUBROUTINE QPLOTZ5

PURPOSE:

Subroutine QPLOTZ5 calculates the scaling information needed to generate the format to label the left-

hand side of the printer plot.

CALLING SEQUENCE:

CALL QPLOTZ5 (PDQ)

ARGUMENT:

PDQ-scaling factor for ordinate plot

COMMON BLOCKS:

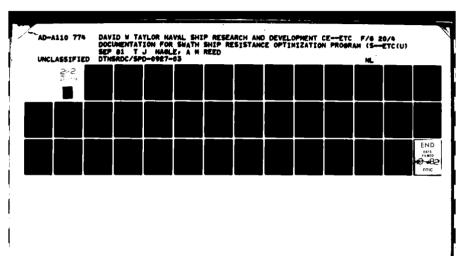
XRPLOTF, XRPLOTQ

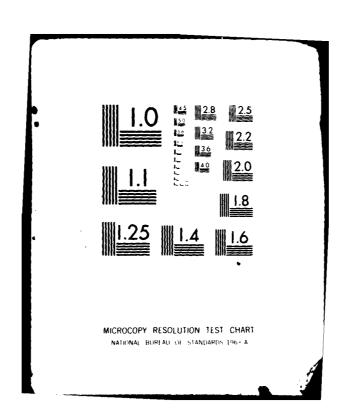
SUBROUTINES CALLED:

NONE

CALLED BY:

PLOT4





FUNCTION RANDUM

PURPOSE:

To avoid resolution valleys, by making random steps on a periodic basis, Function RANDUM returns a random number uniformly distributed in (0,1)

CALLING SEQUENCE:

RA = RANDUM (NAUGHT)

ARGUMENTS:

NAUGHT - number used to initialize

the random number generator

COMMON BLOCKS:

NONE

CALLED BY:

SUBROUTINE REFLKT

PURPOSE:

Subroutine REFLKT defines the lower half of the symmetrical matrices T and W, the auxiliary wave resistance integrals, by reflection about the diagonal of the

matrix.

CALLING SEQUENCE:

CALL REFLKT

ARGUMENTS:

NONE

COMMON BLOCKS:

AUX, COEFS

SUBROUTINE CALLED:

NONE

CALLED BY:

RWAVE

SUBROUTINE RINIT

PURPOSE:

Subroutine RINIT initializes the auxiliary wave resistance matrices, T and W, to zero

CALLING SEQUENCE:

CALL RINIT

ARGUMENT:

NONE

COMMON BLOCKS:

AUX, COEFS

SUBROUTINE CALLED:

NONE

CALLED BY:

RWAVE

SUBROUTING RINTEG

PURPOSE:

Subrouting RINTEG evaluates the integrand for T and W functions.

CALLING SEQUENCE:

CALL RINTEG (ALFA, B, D, NLOC2, WTINT, SEPCOS, SQ)

ARGUMENTS:

ALFA Integrating variable

CSTRT1(I)/XLS CSTRT2(I)/XLS

NLOC2 -Flag indicating the

presence of a second strut

NLOC2 = 0 if single strut

= 1 if tandem strut

WTINT - Weighting constant

for the integrand

SEPCOS - Value of the cosine

function in the

integrand

SQ - Value of the integrand

$$\frac{1}{(\alpha^2-\gamma^2)^{1/2}}$$

COMMON BLOCKS:

AUX, OMEGA, COEFS, WORKVL

SUBROUTINE CALLED:

BESSJ

CALLED BY:

RWAVE

COMMENTS: Integrals used in calculations are of the form:

$$\int_{\gamma_{os}}^{\gamma_{os} + 1} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{os}^2}} f(\alpha) + \int_{\gamma_{os} + 1}^{\infty} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{os}^2}} f(\alpha)$$
or the first integral we use the substitution: $\alpha = 3$

For the first integral we use the substitution: $\alpha = \gamma_{0s} + \zeta^2$ where $d\alpha = 2\zeta d\zeta$.

Therefore the first integral becomes $\int_0^1 \frac{2d\zeta}{\sqrt{\frac{2\gamma_{os}+\zeta^2}{2\gamma_{os}}}} \, f(\gamma_{os}+\zeta^2).$ Where the function $f(\alpha)$ takes the form of $E_s^2(\alpha) \, J_m(\alpha) J_n(\alpha)$

where the order of the Bessel functions is determined by which auxiliary function is to be evaluated.

SUBROUTINE ROUT

PHRPOSE:

Calculates EHP, and wave resistance coefficients, frictional drag coefficients, speed constants, and prints the intermediate and final design

results

CALLING SEQUENCE:

CALL ROUT (TITLE, EHP, KEY)

ARGUMENTS:

TITLE - label printed on output

- effective horsepower of the ship EHP

KEY - variable used to skip or print output

COMMON BLOCKS:

OMEGA, PHYSCO, WORKVL

SUBROUTINES CALLED:

SUM, FORMDR

CALLED BY:

Function EHP

COMMENTS:

The type of output from ROUT is determined by the value of KEY, which is set in the subroutines which call EHP. If KEY = 0 (set in PEHP) no output is printed and ROUT just does computations; if KEY = 1 (set in SWATHO) the optimization is complete and the final design values are printed; if KEY = 2 (set in AUXPLOT) the intermediate design results are printed

SUBROUTINE RWAVE

PURPOSE:

Subroutine RWAVE computes the auxiliary wave

resistance functions T and W.

CALLING SEQUENCE:

CALL RWAVE (B, D, NLOC2)

ARGUMENTS:

B -CSTRT1(I)/XLS D -CSTRT2(I)/XLS

NLOC2 -Flag indicating the presence of a

second strut

 $NLOC2 = \begin{cases} 0 & \text{if single strut} \\ 1 & \text{if tandem struts} \end{cases}$

COMMON BLOCKS:

AUX, OMEGA, PSI, COEFS, PHYSCO, WORKVL

SUBROUTINES CALLED:

RINIT, SIMPSN, RINTEG, BESSJ, REFLKT

CALLED BY:

FUNCTION EHP

COMMENTS:

Integrals evaluated by RWAVE:

$$T_{S_{mn}} = (2m-1) (2n-1) \int_{\gamma_{os}}^{\infty} \frac{d\alpha}{\alpha^2} \sqrt{\alpha^2 - \gamma_{os}^2} E_s^2(\alpha) J_{2m-1}(\alpha) J_{2m-1}(\alpha)$$

$$W_{S_{mn}} = (2m) (2n) \qquad \int_{\gamma_{os}}^{\infty} \frac{d\alpha}{\alpha^2} \sqrt{\alpha^2 - \gamma_{os}^2} \qquad E_{s}^2(\alpha) \qquad J_{2m}(\alpha) \qquad J_{2n}(\alpha)$$

$$T_{B_{mn}} = (2m-1) (2n-1) \int_{\gamma_{os}}^{\infty} \frac{\alpha^2 d\alpha}{\sqrt{\alpha^2 - \gamma_{os}^2}} E_B^2(\beta) J_{2m-1}(\beta) J_{2n-1}(\beta)$$

$$W_{B_{mn}} = (2m) (2n) \qquad \int_{\gamma_{OS}}^{\infty} \frac{\alpha^2 d\alpha}{\sqrt{\alpha^2 - \gamma_{OS}^2}} \qquad E_{B}^{2}(\beta) \quad J_{2n}(\beta)$$

$$T_{SB_{mn}} = (2m-1) (2m-1) \int_{\gamma_{OS}}^{\infty} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{OS}^2}} E_s(\alpha) E_B(\beta) J_{2m-1}(\alpha) J_{2m-1}(\beta)$$

$$W_{SB_{mn}} = (2m) (2n) \qquad \int_{\gamma_{OS}}^{\infty} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{OS}^2}} \qquad E_{S}(\alpha) \quad E_{B}(\beta) \quad J_{2m}(\alpha) \quad J_{2n}(\beta)$$

SUBROUTINE SCALE

PURPOSE:

Creates current working values X, by X = S*XI, where S is the scale factor, and XI is the

vector of initial values.

CALLING SEQUENCE:

CALL SCALE (S, N)

ARGUMENTS:

S - scaling value for each of the variables

N - number of variable

COMMON BLOCK:

INITL, NOCHG, PHYSCO, WORKVL

SUBROUTINES CALLED:

NONE

CALLED BY:

SWATHO, PEHP, AUXPLT

SUBROUTINE SIMPSN

PURPOSE:

Subroutine SIMPSN sets up Simpson's multipliers for numerical integration by Simpson's rule.

CALLING SEQUENCE:

CALL SIMPSN (NPTS, SIMP)

ARGUMENTS:

NPTS - Number of integration steps

SIMP - Array containing the Simpson's multipliers

COMMON BLOCKS:

NONE

SUBROUTINE CALLED:

NONE

CALLED BY:

RWAVE

COMMENT:

These values are used to integrate the auxiliary wave resistance integrals, T and W, for values of the integral between Υ_{OS} and $\Upsilon_{OS}+1$

An example of one of these integrals in the range Y_{os} to $Y_{os}+1$ is

$$\int_{\gamma_{os}}^{\gamma_{os}+1} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{os}^2}} f(\alpha) = \int_0^1 \frac{2d\zeta}{\sqrt{2\gamma_{os} + \zeta^2}} f(\gamma_{os} + \zeta^2)$$

SUBROUTINE SUM

PURPOSE:

Subroutine SUM computes the sums for the wave resistance from the Chebychev coefficients and the auxiliary wave resistance functions, T and W

M N $\Sigma \Sigma (A_m A_n T_m + B_m B_n W_{mn})$ m=1 n=1 m n mn

COMMENTS:

where A_m and B_m are the symmetric and antisymmetric Chebychev coefficients for the strut or body, and T_{mn} and W_{mn} are the auxiliary wave resistance functions for various hull-strut combinations

CALLING SEQUENCE:

CALL SUM (SUM1S, SUM1B, SUM1SB, SUM12S,

SUM12B, SUM12SB)

ARGUMENTS:

SUMIS - Partial sum for strut 1
SUMIB - Partial sum for body 1

SUMISB - Partial sum for interaction between strut 1 and body 1

SUM12S - Partial sum for interaction between strut 1 and strut 2

SUM12SB - Partial sum for interactions between strut 1 and body 2 or

strut 2 and body 1

COMMON BLOCKS:

AUX, COEFS

SUBROUTINE CALLED:

NONE

CALLED BY:

ROUT

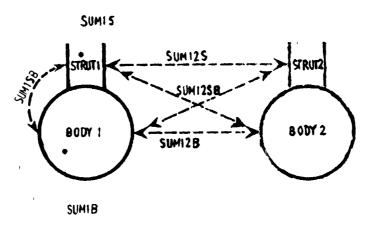


Figure 4 - Body-Strut Configuration

SUBROUTINE VCPRINT

PURPOSE:

With certain input options, this subroutine prints a vector of given length using two of

several formats

CALLING SEQUENCE:

CALL VCPRINT (OPTION, V, N)

ARGUMENTS:

OPTION = 1 Prints the Second Difference Array

= 2 Prints the Scale Factors

= 3 Prints the value of the approximating

quadratic form

= 4 Prints the value of X

V = Matrix of search directions

N = Number of variable

COMMON BLOCKS:

NONE

SUBROUTINES CALLED:

NONE

CALLED BY:

PRAXIS

SUBROUTINE WSURFB

PURPOSE:

Determine surface area of body of revolution whose sectional area is given by

Chebychev coefficients

CALLING SEQUENCE:

CALL WSURFB (AREA)

ARGUMENTS:

AREA - Surface area of a body (ft²)

COMMON BLOCKS:

COEFS, PHYSCO, WORKVL

SUBROUTINES CALLED:

FUNCTION EVAL

CALLED BY:

CHEB

SUBROUTINE WSURFS

PURPOSE:

Determines surface area of a strut whose thick-

ness distribution is given by Chebychev

coefficients

CALLING SEQUENCE:

CALL WSURFS (AREA)

ARGUMENTS:

AREA - Wetted surface area of a strut (ft²)

COMMON BLOCKS:

COEFS, WORKVL

SUBROUTINES CALLED:

FUNCTION EVAL

CALLED BY:

CHEB

FUNCTION YINTP

PURPOSE:

Function YINTP interpolates through a set of discrete data, using quadratic interpolation.

CALLING SEQUENCE:

YTP = YINTP (XZ, X, Y, N)

XA - Point to be interpolated
X - Array of ordinate data
Y - Array of abscissa data
N - Number of data points

COMMON BLOCKS:

NONE

SUBROUTINE CALLED:

NONE

CALLED BY:

FORMDR

COMMENTS:

Uses linear interpolation

REFERENCES

- 1. Lin, W. C. and W. G. Pay, Jr., "The Still Water Resistance and Propulsion Characteristics of Small-Waterplane Area Twin-Hull (SWATH) Ships," AIAA/SNAME Advanced Marine Vehicles Conferences. Paper No. 74-325, (1974).
- 2. Brent, Richard P., "Algorithm for Finding Zeroes and Extrema of Functions without Calculating Derivatives," Stanford University Report, CS-71-198, 1971

APPENDIX A OBJECTIVE AND PENALTY FUNCTIONS

APPENDIX A - OBJECTIVE AND PENALTY FUNCTIONS

The program SWATHO is designed to determine the geometry of a SWATH ship which yields the minimum effective power for a given speed. Thus the objective function is the effective power. However, this is an over simplification of the problem, in that the solution to the problem must also satisfy a number of constraints. That this is so is intuitively obvious if one considers that the ship with minimum power for a given speed is that ship with zero length and volume, and therefore zero resistance. Thus an obvious constraint is that the ship contain at least some minimum volume.

The subroutine PRAXIS, which is used to perform the minimization, is a program designed for use in unconstrained minimization problems. This dictates that a modified objective function be employed, which takes into account the fact that constraints are or are not violated. To meet this requirement, an external constraint method was employed, where the functions $\mathbf{g_i}(\mathbf{x})$ were defined to be greater than zero when the \mathbf{i}^{th} constraint was violated, and less than zero when the \mathbf{i}^{th} constraint was satisfied. Using these $\mathbf{g_i}$'s, the modified objective function was defined as:

PEHP = EHP(1 +
$$\sum_{i}$$
 Max($g_{i}(\underline{x}), 0$)).

In all of the above definitions, the vector $\underline{\mathbf{x}}$ is the vector of parameters defining the ship's geometry.

This definition of the objective function results in the addition of no penalty to the objective function internal to the region where the constraints are satisfied, and results in the addition of a penalty exterior to the region where the constraints are satisfied. This gives rise to the description of this as an external constraint method.

There are a total of nineteen penalty functions which are applied to the problem of minimizing the resistance of a SWATH ship. These constraints/penalty functions are as follows:

1. Minimum displacement constraint

$$g_1 = (DISPMN-DISP)\alpha_1$$

where DISPMN is the minimum ship displacement as input by the user and DISP is the current ship displacement at that iteration of the calculations.

2. Minimum draft-diameter ratio constraint

$$g_2 = \left[\text{Draft Factor x BDIA} - (\text{HS} + \text{BDIA}) \right] \alpha_2$$

where the Draft Factor is the minimum draft to diameter ratio, BDIA is the body diameter at half of the length of the body and HS is the draft of the strut.

3. Maximum draft constraint

$$g_3 = (HS + BDIA - DRFTMAX)\alpha_3$$

where DRFTMAX is the maximum draft of the ship as input by the user.

4. Maximum beam constraint

$$g_{\Delta} = (BS + BDIA - WMAX)\alpha_{\Delta}$$

where BS is the separation of the hull centerlines and WMAX is the maximum beam of the ship as specified by the user.

5. Hull overlap constraint

$$g_5 = (BDIA - BS)\alpha_5$$

6. Strut-hull Nose clearance constraint

$$g_6 = (XFWD + CSTRT + XLS/2 - XLB/2)\alpha_6$$

where XFWD is the minimum distance from the body leading edge to the strut leading edge, CSTRT is the distance of the strut forward of the body centerline, XLS is the length of the strut and XLB is the length of the body.

7. Strut-hull tail clearance constraint

$$g_7 = (XAFT - XLB/2 + XLS/2 - CSTRT)^{\alpha}$$

where XAFT is the minimum distance from body trailing edge to strut trailing edge.

8. Minimum transverse GM constraint

$$g_8 = \left[\text{MINGMT} - (KB + BMT - KG) \right] \alpha_8$$

where MINGMT is the minimum transverse GM, KB is the distance of the center of buoyancy from the baseline, BMT is the transverse metacenter and

KG is the distance of the center of gravity from the baseline.

9. Minimum longitudinal GM contraint

$$g_9 = \left[\text{MINGML} - (KB + BML - KG) \right] \alpha_9$$

where MINGML is the minimum longitudinal GM and BML is the longitudinal metacenter.

10. Body offsets greater than zero contraint

$$g_{10} = \int_{-1}^{1} dx [|b(x)| - b(x)] \alpha_{10}$$

where b(x) represents the body offsets

11. Strut offsets greater than zero constraint

$$g_{11} = \int_{-1}^{1} dx [|t(x)| - t(x)] \alpha_{11}$$

where t(x) represents the strut offsets

12. Maximum length-diameter ratio check

$$g_{12} = (XLB - LODMAX \times BDIA)\alpha_{12}$$

where LODMAX is the maximum body length over diameter ratio.

13. Minimum length-diameter ratio check

$$g_{13} = (LODMIN \times BDIA - XLB)\alpha_{13}$$

where LODMIN is the minimum body length over diameter ratio.

14. Strut minimum thickness constraint

$$g_{14} = (TSMIN - TSMAX)\alpha_{14}$$

where TSMIN is the minimum strut thickness at 50% chord and TSMAX is the strut thickness in the current iteration of the calculations.

15. Strut thickness-body diameter constraint

$$g_{15} = (TSMAX - BDIA)\alpha_{15}$$

16. Positive strut length contraint

$$g_{16} = (-XLS)\alpha_{16}$$

17. Positive body length constraint

$$g_{17} = (-XLB)\alpha_{17}$$

18. Maximum body offset constraint

$$g_{18} = (b_{max} - 1.2 BDIA)\alpha_{18}$$

where b is the maximum body offset. This constraint insures a reasonable value for the maximum body offset.

19. Maximum strut thickness constraint

$$g_{19} = (t_{max} - 1.2 \text{ TSMAX})\alpha_{19}$$

where t_{\max} is the maximum possible value of strut thickness. This constraint insures a reasonable value for the maximum strut thickness.

In all of the above penalty functions, the constraints α_i are chosen so that a constraint violation of ten percent results in a value of g_i around 10^4 .

Of the above penalty functions, eleven are constraints imposed by the program user to insure that the optimum vehicle meets some minimum set of criteria such as minimum displacement and transverse and longitudinal GM. These user specified constraints are the constraints numbered 1, 2, 3, 4, 6, 7, 8, 9, 12, 13, and 14.

The other eight constraints are constraints imposed by the program to insure that the geometry of the vehicle is reasonable. Examples of these would be constraints to insure that the lengths of the body and strut are positive, and constraints to insure that the offsets of the strut and body are all positive. The numbers of these constraints are 5, 10, 11, 15, 16, 17, 18, and 19.

APPENDIX B COMPUTATIONAL PROCEDURE FOR SWATH RESISTANCE PREDICTION

APPENDIX B - COMPUTATIONAL PROCEDURE FOR SWATH RESISTANCE PREDICTION

The main purpose of the SWATHO program is to minimize the power required to propel a SWATH ship in a calm sea at constant speed. For a ship moving at speed V and experiencing a total drag force $R_{\overline{T}}$ the effective horsepower required is:

$$P_{E} = \frac{R_{T}V}{550} .$$

The total drag is comprised of three components: frictional resistance (R_F), wave-making resistance (R_{V}), which form drag (R_{FM}). That is:

$$r_{\rm r} \approx R_{\rm F} + R_{\rm W} + R_{\rm FM}$$

The frictional resistance and form drag are caused by the motion of the hull through a viscous fluid. The wave-making resistance is due to the energy that must be supplied by the ship to the wave system created on the free surface. However, wave-making resistance and form drag are usually grouped together under the title of residuary resistance.

FRICTIONAL RESISTANCE

Frictional resistance is the single largest component of the resistance of a SWATH ship. Frictional resistance is calculated in the traditional fashion of naval architects, using the ITTC model ship correlation line to determine a frictional resistance coefficient (C_F) . However, for a SWATH ship, the frictional drag of the hulls and struts are calculated separately based on their respective Reynolds numbers. Thus we have that the frictional resistance coefficient is given by the formula:

$$C_{F} = \frac{0.075}{(\log_{10} Rn - 2)^{2}}$$

and the frictional resistance of the hulls and struts are given by:

and

In calculating the total frictional resistance, a correlation allowance (C_A) of 0.0005 is also included in the total, so we have that:

$$R_F = R_{Hull} + R_{FStrut} + \frac{1}{2}\rho (S_{Hull} + S_{Strut})V^2 C_A$$

FORM DRAG

Form drag (R_{FM}) is usually defined as the viscous component of the drag due to the shape of body, i.e., the difference between the total viscous drag, and the viscous drag of the equivalent flat plate of the same length. However, in the case of the SWATH resistance programs, the form drag has been determined in a much more empirical fashion.

SWATH form drag has been defined as the difference between the experimentally determined residuary resistance of a hull form and the theoretically derived wave resistance for that same hull form. Such a difference was calculated for both SWATH 3 and SWATH 4 based on bare hull resistance results, 1 and plotted as a function of strut speed-length ratio. A curve was then faired through the data, Figure 3 (p. 69). Due to the scatter in the data a decision was made not to allow the form drag coefficient go below 0.0005; this can be seen on Figure 3. The form drag coefficient, determined from Figure 3, is converted to full scale form drag, using the traditional formula and the total wetted surface:

$$R_{FM} = \frac{1}{2} \rho SV^2 C_{FM}$$
.

WAVE-MAKING RESISTANCE

The wave-making resistance of a ship is not easily predicted from empirical relations or from gross ship features. Experience has shown that two ships of similar form may differ significantly in their measured total resistance due to differences in the wave-making resistance component. Wave-making resistance is a function of the fine details of ship form, and is best studied through mathematical analysis.

Lin and Day investigated the problem of wave-making resistance for twin-hull ships. The mathematical problem is formulated within the context of the linearized thin-ship theory. In their investigations, the potential flow around such a ship is represented by sheet distributions of sources.

The computer program SWATHO documented in this user's manual is devoted primarily to the implementation of numerical procedures for the prediction of the

wave-making resistance of SWATH ships following the theory developed by Lin and Day. The remainder of this appendix will be devoted to the discussion of the essential details involved in the numerical procedure utilized in the wave-making resistance predictions.

Based on the analysis of Lin and Day, the wave resistance is expressed as follows:

$$R_{S} = \left(\frac{\pi}{2} \rho g T^{2} L_{S} \gamma_{os}\right)$$

$$\times \sum_{m=1}^{M} \sum_{n=1}^{M} \left\{A_{Sm} A_{Sn} T_{Smn} + B_{Sm} B_{Sn} W_{Smn}\right\},$$

$$R_{B} = \left(\frac{2\pi \rho g A_{0}^{2}}{L_{S} \gamma_{os}}\right) \sum_{m=1}^{M} \sum_{n=1}^{M} \left\{A_{Bm} A_{Bn} T_{Bmn} + B_{Bm} B_{Bn} W_{SBmn}\right\}$$

$$R_{SB} = (2\pi \rho g A_{0}^{2} T) \sum_{m=1}^{M} \sum_{n=1}^{M} \left\{A_{Sm} A_{Bn} T_{SBmn} + B_{Sm} B_{Bn} W_{SBmn}\right\}$$

where T, L_S , and h_S are the maximum thickness, length, and draft of the strut, respectively; A_0 , L_B , and h_B are the maximum section area, length, and depth of submergence of the axis of the body.

In the above equations, R_S , R_B , and R_{SB} represent the wave resistance due to one strut, one main body, and the intersection between strut and main body, respectively. Hence, the wave resistance of a SWATH ship becomes

$$R_W = 2(R_S + R_B + R_{SB}).$$

Also in the above equations, A_{Sm} and B_{Sm} are the Chebychev coefficients for the strut, A_{Bm} and B_{Bm} are the Chebychev coefficients for the hull, and T_{SMmn} , T_{SBmn} , T_{Bmn} , W_{Smn} , W_{Smn} , and W_{Bmn} are the auxiliary wave resistance functions. The auxiliary wave resistance functions are defined as follows:

$$\frac{\frac{T_{Smn}}{(2m-1)(2n-1)}}{\frac{W_{Smn}}{(2m)(2n)}} = \int_{\gamma_{OS}}^{\infty} \frac{d\alpha}{\alpha^2 \sqrt{\alpha^2 - \gamma_{OS}^2}} p\left(\alpha, \frac{2b}{L_S}, \gamma_{OS}\right) \times E_S^2(\alpha) \left\{ \int_{2m}^{J_{2m-1}(\alpha)} J_{2n-1}(\alpha) \right\}$$

$$D = 1 + \cos \frac{2b}{L_{S}} \frac{2}{\gamma_{os}} \alpha \sqrt{\alpha^{2} - \gamma_{os}^{2}},$$

$$E_{S} = 1 - e^{-2(h_{S}/L_{S})(\alpha^{2}/\gamma_{os})},$$

$$\gamma_{os} = \frac{\varepsilon^{L_{S}}}{(2U^{2})},$$

$$\frac{T_{Bmn}}{(2m-1)(2n-1)}$$

$$\frac{W_{Bmn}}{(2m)(2n)}$$

$$= \int_{\gamma_{os}}^{\infty} d\alpha \frac{\alpha^{2}}{\sqrt{\alpha^{2} - \gamma_{os}^{2}}} D(\alpha, \frac{2b}{L_{S}}, \gamma_{os})$$

$$\times E_{B}(\beta) \begin{cases} J_{2m-1}(\beta) & J_{2n-1}(\beta) \\ J_{2m}(\beta) & J_{2n}(\beta) \end{cases}$$

$$E_{B} = e^{-2(h_{B}/L_{B})(\beta^{2}/\gamma_{OB})},$$

$$\gamma_{OB} = \frac{gL_{B}}{(2U^{2})},$$

$$\beta = (\frac{\gamma_{OB}}{\gamma_{os}}) \alpha,$$

and

$$\frac{T_{SBmn}}{(2m-1)(2n-1)} = \int_{\gamma_{OS}}^{\infty} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{OS}}} D(\alpha, \frac{2b}{L_S}, \gamma_{OS})$$

$$\times E_{S}(\alpha) E_{B}(\beta) \begin{cases} J_{2m-1}(\alpha) J_{2n-1}(\beta) \\ J_{2m}(\alpha) J_{2n}(\beta) \end{cases}$$

where $J_m(\alpha)$ is a Bessel function of the first kind, defined as follows:

$$J_{m}(\alpha) = \frac{2}{\pi} \int_{0}^{\pi/2} \cos(\alpha \sin t) \cos(mt) dt.$$

NUMERICAL INTEGRATION METHODS FOR AUXILIARY WAVE RESISTANCE FUNCTION EVALUATION

The range of integration for the auxiliary wave resistance functions is divided into three intervals of integration: $\{\gamma_{os}, \gamma_{os} + 1\}$, a "central region" and the "tail" region. A detailed discussion of these intervals and the integration procedure for each interval follows.

INTEGRATION PROCEDURE FOR THE INTERVAL $[\gamma_{os}, \gamma_{os} + 1]$

In the region of γ_{OS} the integrands do not behave well, but they are nevertheless integrable. This is due to the presence of $\sqrt{\alpha^2-\gamma_{OS}^2}$ in the denominator. This may be easily handled by utilizing the substitution:

$$\alpha = \gamma_{os} + \zeta^2$$

where

$$d\alpha = 2\zeta d\zeta$$
.

This substitution will be utilized in the region $[\gamma_{os}, \gamma_{os} + 1]$. It eliminates the square-root singularity by giving

$$\int_{\gamma_{os}}^{\gamma_{os+1}} \frac{d\alpha}{\sqrt{\alpha^2 - \gamma_{os}^2}} f(\alpha) = \int_0^1 \frac{2d\zeta}{\sqrt{2\gamma_{os} + \zeta^2}} f(\gamma_{os} + \zeta^2).$$

Investigating the behavior of the integrands as functions of ζ in the region $0 \le \zeta \le 1$ shows that they are very well behaved and a simple numerical integration scheme with about 30 points will yield excellent results.

INTEGRATION PROCEDURE FOR THE CENTRAL REGION

The central region of the numerical integration starts at α = γ + 1 and continues until $E_R(\beta)$ reaches a very small fraction of its initial value,

$$E_B(\beta_{init}) = E_B \left[\frac{(\gamma_{os} + 1)\gamma_{OB}}{\gamma_{os}} \right].$$

The fraction is given as $10^{-\epsilon}$, so

$$\frac{\exp\left\{-2\frac{h_B}{L_B}\frac{\beta max^2}{\gamma_{OB}}\right\}}{\exp\left\{-2\frac{h_B}{L_B}\frac{(\gamma_{OS}+1)^2\gamma_{OB}}{\gamma_{OS}^2\gamma_{OB}}\right\}} = 10^{-\varepsilon} = \exp(-2.3026\varepsilon).$$

Hence,

$$\alpha_{\text{max}} = \frac{L_{\text{S}}}{L_{\text{B}}} \beta_{\text{max}} = \sqrt{\left(\frac{2.3026\epsilon\gamma_{\text{os}}}{\frac{h_{\text{B}}}{L_{\text{B}}} \frac{L_{\text{B}}}{L_{\text{S}}}}\right)} + (\gamma_{\text{os}} + 1)^{2}.$$

The justification of this choice for α_{max} , and the effect of the choice of ϵ , will be discussed in the next section.

In order to implement an effective and economical numerical integration scheme, it is necessary to investigate the behavior of the integrands within this central region. The requirement is to provide at least several points of the integrand for each full cycle of its oscillation. Hence, a conservative estimate of behavior will be an estimate of more rapid oscillation than actually exists.

The Bessel functions are approximated as

$$J_{\nu}(z) \approx \sqrt{\frac{2}{\pi z}} \cos \left(z - \frac{\nu \pi}{2} - \frac{\pi}{4}\right)$$

for |Z| >> 1 and |Z| >> |v|.

for smaller Z, the behavior will be less oscillatory, so this represents a conservative estimate for all Z. This assumption results in

$$J_{\mu}(\alpha)J_{\nu}(\alpha) \approx \begin{cases} \frac{1}{\pi\alpha} \left[\cos(2\alpha+\Psi) + 1\right] & \frac{\mu-\nu}{2} \text{ even} \\ \\ \frac{1}{\pi\alpha} \left[\cos(2\alpha+\Psi) - 1\right] & \frac{\mu-\nu}{2} \text{ odd} \end{cases}$$

where the case of $(\mu-\nu)$ odd will never occur due to the form of the integrands. The other combinations of Bessel functions are found as

$$J_{m}(\alpha)J_{n}(\beta) \sim \frac{1}{\pi\sqrt{\alpha\beta}}\cos(\alpha+\beta)$$

$$J_{\mathbf{m}}(\beta)J_{\mathbf{n}}(\beta) \sim \frac{1}{\pi\beta}\cos(2\beta)$$
,

where the phase angles, a slow oscillation in the first <u>case</u>, and a constant term in the second <u>case</u>, have all been ignored. Since β is generally greater than α , but of the same order of magnitude, a conservative estimate gives the variation of all of the above as

$$J_{m}J_{n} \sim \frac{1}{\pi\alpha} \cos(2\beta) = \frac{1}{\pi\alpha} \cos\left(\frac{2L_{B}}{L_{S}}\alpha\right).$$

The term $[\sqrt{\alpha^2-\gamma_{os}^2}]$ behaves as $[\alpha]$ for $\alpha >> \gamma_{os}$, and is better behaved for smaller α . This conservative estimate will be used for the occurrence of this term at the beginning of each integrand. However, the substitution will not be made in the term

$$D(\alpha) = \cos \left[\frac{2}{\gamma_{os}} \frac{2b}{L_{S}} \alpha \sqrt{\alpha^{2} - \gamma_{os}^{2}} \right],$$

which occurs in all of the two-hull integrands, since that would result in considerably less economical numerical integration.

The terms $E_B(\beta)$ and $E_S(\alpha)$ contain decreasing exponentials and are not oscillatory and are quite well-behaved, and so will not affect the spacing of points for the numerical integration.

Therefore, the integrand for $T_{12S_{mn}}$ and $W_{12S_{mn}}$ behave like

$$\frac{d\alpha}{\pi\alpha^4} \ E_S^2(\alpha) \ \cos\left(2 \ \frac{L_B}{L_S} \ \alpha\right) \ \cos\left[\ \frac{2}{\gamma_{os}} \ \frac{2b}{L_S} \ \alpha\sqrt{\alpha^2-\gamma_{os}^2} \ \right],$$

those for $T_{12B_{mn}}$ and $W_{12B_{mn}}$ behave like

$$\frac{d\alpha}{\pi} E_B^2 \left(\frac{L_B}{L_S} \alpha \right) \cos \left(2 \frac{L_B}{L_S} \alpha \right) \cos \left[\frac{2}{\gamma_{os}} \frac{2b}{L_S} \alpha \sqrt{\alpha^2 - \gamma_{os}^2} \right], \text{ and}$$

those for ${\rm T_{12SB_{mn}}}$ and ${\rm W_{12SB_{mn}}}$ behave like

$$\frac{d\alpha}{\pi\alpha^2} \ E_B \bigg(\frac{L_B}{L_S} \ \alpha \bigg) E_S (\alpha) \ \cos \bigg(2 \ \frac{L_B}{L_S} \ \alpha \bigg) \ \cos \left[\ \frac{2}{\gamma_{os}} \ \frac{2b}{L_S} \ \alpha \sqrt{\alpha^2 - \gamma_{os}^2} \ \right] \ .$$

The single-hull terms (T_{Smn}, W_{Smn}, T_{Bmn}, W_{Bmn}, T_{SBmn}, W_{SBmn}) are similar but without the cos $\left[\frac{2}{\gamma_{os}} \frac{2b}{L_S} \alpha \sqrt{\alpha^2 - \gamma_{os}^2}\right]$ term.

Based on the behavior shown above, the most rapid oscillation of the integrands is like

$$\cos \left[2 \; \frac{L_B}{L_S} \; \alpha \right] \cos \frac{2}{\gamma_{os}} \! \left[\! \frac{2b}{L_S} \; \! \alpha \sqrt{\alpha^2 \! - \! \gamma_{os}^2} \right] \, . \label{eq:cos}$$

In the vicinity of $\alpha = \overline{\alpha}$, $\cos[f(\alpha)]$ behaves like $\cos\left[\left(\frac{\mathrm{d}f(\alpha)}{\mathrm{d}\alpha}\right)\right]$ and hence

$$\cos \left[2 \, \frac{L_{B}}{L_{S}} \, \alpha \right] \, \cos \left[\frac{2}{\gamma_{os}} \, \frac{2b}{L_{S}} \, \sqrt{\overline{\alpha}^{2} - \gamma_{os}^{2}} \right. + \frac{\overline{\alpha}^{2}}{\sqrt{\overline{\alpha}^{2} - \gamma_{os}^{2}}} \, \alpha \right]$$

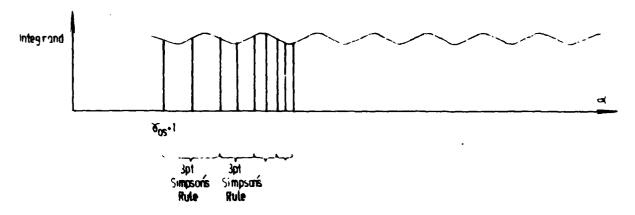
$$\approx \cos \left[\left(2 \frac{L_{B}}{L_{S}} + \frac{2}{\gamma_{os}} \frac{2b}{L_{S}} \frac{2\overline{\alpha}^{2} - \gamma_{os}^{2}}{\sqrt{\overline{\alpha}^{2} - \gamma_{os}^{2}}} \right) \alpha \right].$$

If the accuracy desired from the numerical integration scheme requires P points per cycle, then the step size S (distance between points) in the vicinity of $\overline{\alpha} = \alpha$ is

$$S = \begin{bmatrix} \frac{2\pi}{P} \\ \frac{L_{B}}{2 \frac{L_{B}}{L_{S}} + \frac{2}{\gamma_{os}} \frac{2b}{L_{S}} \frac{2\overline{\alpha}^{2} - \gamma_{os}^{2}}{\sqrt{\overline{\alpha}^{2} - \gamma_{os}^{2}}} \end{bmatrix}.$$

This step size will decrease with increasing α but will be smaller for larger hull centerline spacing b. Hence, if computations are made for various values of b for the same sample points, the step size should be based on the largest value of b.

The numerical integration scheme chosen for the central region is a modified Simpson's rule in which all steps are not equal. For this method, a step size is computed at a given α (starting first with $\gamma_{os}+1$), using the last equation. Then an integral is taken over an interval equal to two of these step sizes, using a three-point Simpson's rule. A new step size is computed for the α at the end of that interval, and the process is repeated until α_{max} is reached or exceeded. The concept may be visualized as shown below.



This economizes on samples (points) where the integrand does not vary quickly, and uses more densely spaced points where necessary.

All of the approximations made in this section have been only for the purpose of choosing the step sizes for the numerical integration. The samples taken, which are multiplied by appropriate weights added to obtain the numerical evaluation of the integral, are all found using the actual integrands.

INTEGRATION PROCEDURE FOR THE TAIL

It should be noted that the criterion for the end of the "central region," and hence the beginning of the "tail," is that $E_B(\beta)$ reach a very small percentage of its original value. This will insure that any integrand which includes $E(\beta)$ will have long ceased to contribute significantly to the result. The body-alone and strut-body integrals $(T_B, W_B, T_{12B}, W_{12B}, T_{SB}, W_{SB}, T_{12SB}$ and W_{12SB} can therefore be assumed complete at the conclusion of the "central region" numerical integration.

The strut-alone terms (T_S , W_S , T_{12S} , and W_{12S}) are not complete, however. The integrand for these terms decreases only like α^{-4} . The utilization of the "central region" numerical integration scheme until this integrand were very small would be quite accurate but exceptionally expensive, especially due to the constantly decreasing (with increasing α) step size required for that scheme. Hence, a more economical scheme must be developed, accepting the resulting loss of accuracy.

The single-hull strut-alone terms (T $_{\rm S}$ and W $_{\rm S}$) are easily handled since they do not include the highly oscillatory term

$$D(\alpha) = \cos \left[\frac{2}{\gamma_{os}} \frac{2b}{L_{s}} \alpha \sqrt{\alpha^{2} - \gamma_{os}^{2}} \right].$$

The behavior of the integrands for these terms is like

$$\frac{\mathrm{d}\alpha}{\pi\alpha^4} E_{\mathrm{S}}^2(\alpha) \cos(2\alpha),$$

which allows the use of a simple numerical integration scheme with constant step size

$$S = \frac{\pi}{P}$$

where P is the desired number of points per cycle.

The above method is still not very economical due to the large region over which the integral must be taken, due to the relatively slowly decreasing α^{-4} envelope in which the integrand oscillates. A more economical method is to approximate the integrand asymptotically, and analytically integrate the approximate integrand from a given value to infinity. The obvious simplification is to assume that $E_S(\alpha)\approx 1$, which is an excellent approximation, because $E_S(\alpha)$ approaches unity as $E_B(\beta)$ approaches zero. The approximation is hence assured by the condition that

determined the end of the "central region." In addition to $E_S(\alpha)$, the Bessel functions must also be approximated. We use the approximation

$$J_{\nu}(a) \approx \sqrt{\frac{2}{\pi\alpha}} \cos\left(2 - \frac{\nu\pi}{2} - \frac{\pi}{4}\right)$$

for
$$|\alpha| \gg 1$$
 and $|\alpha| \gg |\nu|$.

If $|\alpha|$ is not enough greater than $|\nu|$ at the beginning of the tail, a special "base of tail" numerical integration is done for T_S and W_S using step size $S = \pi/P$ (as discussed on the previous page). This is done over a region from the start of the "tail" until α is large enough for the above asymptotic approximation to yield an accurate result. The above asymptotic approximation gives

$$J\mu(\nu)J\nu(\mu) \approx \begin{cases} \frac{1}{\pi\alpha} \left[\cos(2\alpha+\Psi) + 1\right] & \frac{\mu-\nu}{2} \text{ even} \\ \frac{1}{\pi\alpha} \left[\cos(2\alpha+\Psi) - 1\right] & \frac{\mu-\nu}{2} \text{ odd} \end{cases}$$

where the case of $(\mu-\nu)$ odd will never occur due to the form of the integrands. The integrands therefore consist of a constant term and an oscillatory term. The contribution of the constant term to the integral, using the above approximations, is

$$\int_{\alpha_{A}}^{\infty} \frac{d\alpha}{\pi \alpha^{3} \sqrt{\alpha^{2} - \gamma_{OS}^{2}}} = \frac{1}{2\pi \gamma_{OS}^{3}} \operatorname{arc sin}\left(\frac{\gamma_{OS}}{\alpha_{A}}\right) - \frac{\sqrt{\alpha_{A}^{2} - \gamma_{OS}^{2}}}{2\pi \gamma_{OS}^{2} \alpha_{A}^{2}}.$$

When α_A is much greater than γ_{os} , this expression will not provide an accurate solution since it will be the difference of two very similar values. For $R = \gamma_{os}/\alpha_A >> 1$, the result is

$$\frac{1}{2\pi\gamma_{OS}^3} \left[R + \frac{1}{6} R^3 + \frac{3}{40} R^5 + \dots \right] - \frac{1}{2\pi\gamma_{OS}^3} \left[R - \frac{1}{2} R^3 - \frac{1}{8} R^5 + \dots \right]$$

$$= \frac{1}{3\pi\alpha_A^3} \left[1 + \frac{3}{10} R^2 + \dots \right].$$

The simplified form

 $\frac{1}{3\pi\alpha_A^3}$

will therefore be accurate to about 2n decimal places if R is 10^{-n} . Likewise, the exact solution will lose 2n decimal places in the subtraction of the two parts when R is 10^{-n} . Hence, for a computer with about 16 decimal place accuracy, the simplified form is used whenever R < 10^{-4} .

The analytic integration of part of the integrands for T_S and W_S shown in the preceding paragraph depends on a large enough α_A for sufficiently accurate approximation of the Bessel functions. Hence, the simple numerical integration shown must be utilized to fill in the gap if the "central region" ends before such a sufficiently large α_A is reached.

The oscillatory components of the integrands for T_S and W_S in the region of analytic integration of the asymptotic approximations have been ignored in this formulation, as their contribution is assumed to be negligible.

The integrals for T_{12S} and W_{12S} were not carried past the end of the "central region" due to their highly oscillatory integrands, which made the contribution from further computation both uneconomical, and not critical to the final values of T_{12S} and W_{12S} . This is, however, not as good an approximation as those previously noted in the above development. As a result, errors of as much as 2% can be expected in the computation of wave drag due to interference between the two struts.

APPENDIX C

RELATIONSHIPS BETWEEN CHEBYCHEV SERIES AND SWATH HULLFORM COEFFICIENTS

APPENDIX C - RELATIONSHIPS BETWEEN CHEBYCHEV SERIES AND SWATH HULLFORM COEFFICIENTS

CHEBYCHEV SERIES

In the wave resistance integrals, geometry of strut and body are represented by a special form of Chebychev series. In SWATHO, the Chebychev coefficients are approximated from various statistical moments of strut and body. The Chebychev series representation of hull and body geometries are discussed below. Let

$$x = \sin \theta, -\frac{\pi}{2} \le \theta \le \frac{\pi}{2}$$
.

Define the fundamental functions of the Chebychev series as follows:

$$U_{m}(x) = \cos(2m-1)\theta$$
$$= \cos[(2m-1)\sin^{-1}x]$$

$$V_{m}(x) = \sin 2m\theta$$

= $\sin[2m \sin^{-1}x]$, m = 1, 2, . . . , m.

The strut half-thickness and the body sectional-area functions can be represented by finite sums of the fundamental Chebychev series as follows:

$$t(x) = \sum_{m=1}^{M} \left[A_{sm}U_{m}(x) + B_{sm}V_{m}(x) \right]$$
$$= \sum_{m=1}^{M} \left[A_{sm}\cos(2m-1)\theta + B_{sm}\sin 2m\theta \right]$$

and

$$A(x) = \sum_{m=1}^{M} \left[A_{bm}U_{m}(x) + B_{bm}V_{m}(x) \right]$$

$$= \sum_{m=1}^{M} \left[A_{bm}\cos(2m-1)\theta + B_{bm}\sin 2m\theta \right]$$

Through use of the orthogonality of the series, the following inversion formulas are obtained:

$$A_{Sm} = \frac{2}{\pi} \int_{-1}^{1} dx \frac{t(x)U_{m}(x)}{\sqrt{1-x^{2}}}$$

$$= \frac{2}{\pi} \int_{-\pi/2}^{\pi/2} d\theta t(\sin\theta) \cos[(2m-1)\theta],$$

$$B_{Sm} = \frac{2}{\pi} \int_{-1}^{1} dx \frac{t(x)V_{m}(x)}{\sqrt{1-x^{2}}}$$

$$= \frac{2}{\pi} \int_{-\pi/2}^{\pi/2} d\theta t (\sin \theta) \sin(2m\theta),$$

$$A_{Bm} = \frac{2}{\pi} \int_{-\pi/2}^{\pi/2} d\theta \ A(\sin \theta) \cos(2m-1)\theta,$$

and

$$B_{Bm} = \frac{2}{\pi} \int_{-\pi/2}^{\pi/2} d\theta \ A(\sin\theta) \sin 2m\theta.$$

INVERSE PROCEDURE FOR DETERMINING THE CHEBYCHEV COEFFICIENTS

The geometric coefficients of a SWATH ship's form can be shown to be closely related to the Chebychev coefficients of the strut and body.

These coefficients of form are easily obtained at the preliminary design stage and, thus, can be used to approximate the Chebychev coefficients. In the following example, the waterplane coefficient, $C_{\overline{WP}}$, can be used to find $A_{\overline{S1}}$.

By definition, the waterplane area is

$$A_W = 2 \int_{-L_S/2}^{L_S/2} dx \ t(x)$$

where t(x) is the thickness function of the strut. Let

$$x = \frac{L_s}{2} \xi$$
, $dx = \frac{L_s}{2} d\xi$, $\xi = \frac{1}{L_s/2} x$, $t(x) = tmax \cdot t(\xi)$.

Thus,

$$A_{W} = \frac{L_{s}}{2} t_{max} \int_{-1}^{1} d\xi t(\xi)$$

$$= \frac{L_{s}}{2} t_{max} \int_{-1}^{1} d\xi \left[\sum_{1}^{M} A_{sm} U_{m}(\xi) + B_{sm} V_{m}(\xi) \right]$$

$$= \frac{L_{s}}{2} t_{max} \left[\sum_{1}^{M} A_{sm} \int_{-1}^{1} U_{m}(\xi) d\xi + \sum_{1}^{M} B_{sm} \int_{-1}^{1} V_{m}(\xi) d\xi \right].$$

Making the substitution,

$$x = \sin \theta$$
, $dx = \cos \theta d\theta$,

we obtain:

$$A_{W} = \frac{L_{s}}{2} t_{max} \left[\sum_{1}^{M} A_{sm} \int_{-\pi/2}^{\pi/2} d\theta \cos(2m-1)\theta \cos\theta + \sum_{1}^{M} B_{sm} \int_{-\pi/2}^{\pi/2} d\theta \sin 2m\theta \cos\theta \right].$$

The second integral is odd and, therefore, it is identically zero. Hence

$$A_{W} = \frac{L_{st_{max}}}{2} \sum_{1}^{M} A_{sm} \int_{-\pi/2}^{\pi/2} d\theta [\cos(2m-1)\theta \cos\theta]$$

$$= \frac{L_{st_{max}}}{2} \sum_{1}^{M} A_{sm} \int_{-\pi/2}^{\pi/2} d\theta [\cos(2m-2)\theta + \cos 2m\theta]$$

$$= \frac{L_{s}t_{max}}{2} \sum_{1}^{M} A_{sm} \left[\frac{1}{2m-2} \sin(2m-2)\theta + \frac{1}{2m} \sin 2m\theta \right]_{0}^{\pi/2}$$

$$= \frac{L_{s}t_{max}}{2} \sum_{1}^{M} A_{sm} \left[\frac{\sin(m-1)\pi}{2(m-1)} + \frac{\sin m\pi}{2m} \right].$$

These integrals are identically zero except for m=1 where the first term contributes $\pi/2$,

$$A_{W} = \frac{L_{st_{max}}}{2} \frac{\pi}{2} A_{S1}.$$

The waterplane coefficient is defined as

$$C_{WP} = \frac{A_W}{L_{st_{max}}}.$$

Therefore

$$A_{S1} = \frac{4C_{WP}}{\pi} .$$

This derivation shows how the waterplane coefficient is related to the Chebychev coefficient, ${\rm A}_{\rm S1}$. Similarly we can prove,

$$A_{B1} = \frac{4C_{p}}{\pi}$$

where $C_{\mathbf{p}}$ is the body prismatic coefficient.

In the SWATHO program the maximum number of terms of the Chebychev series is three. The Chebychev coefficients are determined by the following formulas for the strut:

$$A_{S1} = \frac{4C_{WP}}{\pi} ,$$

$$A_{S2} = A_{S1}(1 - 16 C_{IW}),$$

$$A_{S3} = 1 - A_{S1} - A_{S2},$$
 $B_{S1} = 4C_{LCF} \cdot A_{S1},$
 $B_{S2} = 0,$
 $B_{S3} = 0,$

and the body coefficients are determined as follows:

$$A_{B1} = \frac{4}{\pi} C_{P},$$
 $A_{B2} = 1 - A_{B1},$
 $A_{B3} = 0,$
 $B_{B1} = 4C_{LCF} \cdot A_{B1},$
 $B_{B2} = 0,$
 $B_{B3} = 0,$

where

 C_{WP} = Waterplane area coefficient C_{WP} = $A_W/(L_S \cdot T_S)$, where A_W is waterplane area of one strut.

 C_{LCF} = Waterplane moment coefficient $C_{LCF} = M_{X_S}/(A_W \cdot L_S)^*$

 C_{IW} = Waterplane inertia coefficient $C_{IW} = I_{WX}/(A_W \cdot L_S)^2$

 C_p = Body prismatic coefficient $C_p = \nabla_B/(A_X \cdot L_B) \text{ where } \nabla_B \text{ is displaced volume of one body.}$

 $C_{LCB} = Body moment coefficient$ $<math>M_{XB}/(\nabla_B \cdot L_B)^*$

All moments and the moment of inertia are taken about the mid-length of the respective strut or body.

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